Various aircraft routing options for air traffic simulation in the chemistry-climate model EMAC 2.53: AirTraf 2.0

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Abstract. Climate impact of aviation is expected to increase further. Aircraft routings are an important measure for climate impact reductions. To find an effective aircraft routing strategy for reducing the impact, the first version of the submodel AirTraf has been developed; this submodel can simulate global air traffic in the ECHAM/MESSy Atmospheric Chemistry (EMAC) model. This paper describes the updated submodel AirTraf 2.0. Seven new aircraft routing options are introduced, including contrail avoidance, minimum economic costs, and minimum climate impact. Example simulations of the new routing options are presented by using around 100 north-Atlantic flights of an Airbus A330 aircraft for a typical winter day. The results clearly show that the family of optimum flight trajectories (three-dimensional) varies according to the routing options. The comparison of the results for various routing options reveals characteristics of the routing with respect to air traffic performances. The minimum cost option obtains a trade-off solution between the minimum time and the minimum fuel solutions. The aircraft routings for contrail avoidance and minimum climate impact reduce the potential climate impact, which is estimated by using algorithmic Climate Change Functions, whereas these two routings increase flight operating costs. A trade-off between the aircraft operating costs and the climate impact is confirmed. The simulation results are compared with literature data and the consistency of the submodel AirTraf 2.0 is verified.

1 Introduction

Climate impact due to aviation emissions is an important issue. Nowadays the global aviation contributes about 5 % to the anthropogenic climate impact (Skeie et al., 2009; Lee et al., 2009, 2010). The aviation is not the largest contributor to climate impact at the moment (e.g., the road transport contributes 11 % to the anthropogenic climate impact; Skeie et al., 2009). However, the aviation's contribution to climate impact is expected to increase further, because global air traffic strongly grows in terms of Revenue Passenger Kilometres (RPK) by 7.4 % in 2016 compared to 2015 (ICAO, 2017). The aviation climate impact consists of carbon dioxide (CO_2) emissions and of non-CO_2 effects. The non-CO_2 effects comprise nitrogen oxides (NO_x) leading to concentration changes of ozone and methane, water vapor (H_2O), hydrocarbons (HC), carbon monoxide (CO), sulfur oxides (SO_x), nonvolatile black carbon (BC or soot), persistent linear contrails, and contrail-induced cirrus clouds (Wuebbles et al., 2007; Lee et al., 2009). These effects change the radiative balance of the Earth's climate system and cause

a radiative impact (RI). The RI potentially drives the climate system into a new state of equilibrium through temperature changes. Lee et al. (2009) stated that the CO_2 emission has the main impact and that the estimated RI of aviation CO_2 in 2005 was $28.0 \, \mathrm{mWm^{-2}}$ ($15.2-40.8 \, \mathrm{mWm^{-2}}$, 90 % likelihood range). The non-CO_2 emissions and the induced clouds also have a large effect on RIs; for example, the estimated RIs in 2005 for total NO_x and for persistent linear contrails were $12.6 \, \mathrm{mWm^{-2}}$

5   ($3.8-15.7 \, \mathrm{mWm^{-2}}$, 90 % likelihood range) and $11.8 \, \mathrm{mWm^{-2}}$ ($5.4-25.6 \, \mathrm{mWm^{-2}}$, 90 % likelihood range), respectively (Lee et al., 2009). Here, the difference between time scales of their impacts is noted. The emitted CO_2 has a long residence time (a century) and becomes uniformly mixed in the whole atmosphere, i.e., its impact is proportional to fuel use. In contrast, the non-CO_2 effects occur on short time scales, e.g., the emitted NO_x remains for a few days to months; the contrails last several hours. Thus, the non-CO_2 effects depend strongly on the ambient (local) atmospheric conditions (Fichter et al., 2005;

10 Mannstein et al., 2005; Gauss et al., 2006; Grewe and Stenke, 2008; Frömming et al., 2012; Brasseur et al., 2016). To investigate measures for reducing the aviation climate impact, the impact of both, CO_2 and non-CO_2 effects, must be considered; therefore, geographic location, altitude, the time of released non-CO_2 emissions and induced clouds, and corresponding local atmospheric conditions need to be considered.

In recent years, Grewe et al. (2017a, b) and Matthes et al. (2012, 2017) have proposed a climate-optimized routing as an

15 important operational measure for reducing the aviation climate impact. This routing allows a significant reduction of the climate impact by optimizing flight routes to avoid regions, where released emissions (including contrails) have a large climate impact. The climate-optimized routing is applicable to present airline fleets. Moreover, the routing can be used in addition to technological measures for reducing the aviation climate impact (e.g., efficient engines, blended wing-body configurations, and laminar flow controls; Green, 2005).

20 Benefits of the climate-optimized routing have been examined before (Gierens et al., 2008; Schumann et al., 2011; Sridhar et al., 2013; Søvde et al., 2014); for example, Frömming et al. (2013) and Grewe et al. (2014b) developed Climate Cost Functions (CCFs) for the climate-optimized routing. The CCFs can identify climate sensitive regions with respect to aviation's CO_2 and non-CO_2 effects (H_2O, ozone, methane, primary mode ozone, and contrails) and estimate corresponding climate impacts. Here, ozone changes arizen from changes of methane are called primary mode ozone (Dahlmann et al., 2016). Grewe et al.

25 (2014a) calculated the CCFs for a winter day and optimized one-day trans-Atlantic air traffic (391 eastbound and 394 westbound flights) using the CCFs in the the system for traffic assignment and analysis at macroscopic level (SAAM; Eurocontrol, 2012). They reported that the climate impact decreased by up to 25 % with a small increase in economic costs of less than 0.5 %. This revealed a great potential for the climate-optimized routing. On the other hand, a trade-off between climate impact and economic cost existed, i.e., the climate-optimized and the cost-optimized routings were conflicting strategies. Grewe et al.

30 (2017b) extended this study and investigated the feasibility of the climate-optimized routing for realistic conditions. Similar trans-Atlantic air traffic simulations (about 800 flights) were performed for five representative winter and three representative summer days, taking safety aspects into account. They found that a decrease in potential climate impact of 10 % was achieved by a cost increase of only 1 %.

Lührs et al. (2016) performed a flight trajectory optimization for nine sample trans-Atlantic routes for a specific weather

35 pattern in winter by the Trajectory Optimization Module (TOM). The trajectories were optimized for economic cost (expressed

by the cash operating cost (COC; Liebeck et al., 1995; see Sect. 2.5.6), which is commonly used as a criterion for airline economics) and for climate impact (measured as average temperature response estimated by integrating the CCFs). The results showed that the climate-optimal route differed from the cost-optimal route. The climate-optimum trajectory (3D-optimized trajectory in lateral and vertical) decreased the climate impact by about 45 % over that of the economical route, whereas it

5 increased COC by 2 %. Thus, the climate impact drastically decreased with a small increase of economic cost.

In general, benefits of flying climate-optimized trajectories were investigated by using different climate metrics. Ng et al. (2014) optimized flight trajectories for a total climate cost which was calculated by the absolute global temperature change potential (pulse AGTP values for three different time horizons; Shine et al., 2005) due to CO_2 emission and contrails. A total of 960 trans-Atlantic flights (482 eastbound and 478 westbound flights) was analyzed for a specific summer day. They reported

10 that the climate-optimized routing reduced the total AGTP (for the medium-term climate goal, i.e., the time horizon of 50 years) by 38 % with an additional flight time of 3.1 % and with extra fuel use of 3.1 % for the eastbound flights, whereas the routing reduced the total AGTP by 20 % with an additional flight time of 3.0 % and with extra fuel use of 3.7 % for the westbound flights. Generally, aircraft operating costs depend on time and on fuel. Thus, those results indicate the aforementioned trade-off between climate impact and economic cost. Although many studies show the benefit of the climate-optimized routing, this

15 routing is not used for the today's flight planning: the today's aircraft routing focuses on the minimum economic cost. However, if additional costs, such as environmental taxes, for aviation climate impact of CO_2 and non-CO_2 effects are included in the operating costs, a cost increase due to the climate-optimized routing is possibly compensated (Grewe et al., 2017b). This inclusion can change the current routing strategy, and incentivize airlines to introduce a climate-optimized flight planning.

Here, we present an air traffic simulation model which serves as a basis for the following ultimate two aims: to investigate

20 an eco-efficient aircraft routing strategy that reduces the climate impact of global air traffic over the next few decades, and to estimate its mitigation gain for different aircraft routing strategies. For these aims, the submodel AirTraf (version 1.0) has been developed as one of the submodels of the ECHAM/MESSy Atmospheric Chemistry (EMAC) model (Yamashita et al., 2015, 2016). AirTraf can simulate global air traffic in EMAC (online) for various aircraft routing strategies (options). Every flight trajectory is optimized for a selected routing option under daily changing local atmospheric conditions. AirTraf can take into

25 account where and when aviation emissions are released or contrails form. The road map for our overall study has been shown elsewhere (Grewe et al., 2017b; Matthes et al., 2017).

This paper presents a technical description of AirTraf (version 2.0). The simple aircraft routing options of great circle (minimum flight distance) and flight time (minimum time) were developed in AirTraf 1.0 (Yamashita et al., 2016). Here, seven new aircraft routing options have been introduced: fuel use, NO_x emission, H_2O emission, contrail formation, simple operating

30 cost (SOC), COC, and climate impact estimated by the algorithmic Climate Change Functions (aCCFs; Van Manen, 2017; Yin et al., 2018a; Van Manen and Grewe, 2019; Yin et al. (manuscript in preparation, 2019); the Climate Change Functions were previously referred to as the Climate Cost Functions mentioned above). These options represent the objects to be minimized. Overall the nine options have been integrated in AirTraf 2.0, which enable air traffic simulations for the ultimate aims of our study (hereinafter the aircraft routing options are referred to simply as, e.g. the 'fuel option'). Thus, the development

35 described in this paper is an indispensable update. Moreover, this paper provides example applications of AirTraf 2.0. Some

simulations of the nine routing options were carried out for trans-Atlantic routes for a typical winter day. Optimum flight trajectories and characteristics of the routing options were analyzed.

Here, we focus on the variety of the routing options. Various routing options have been made available in AirTraf, because not only the climate and the cost options, but also the other options are important subjects for air traffic routing studies. The time option is useful for delay recovery. Delays cause costs to airlines. Thus, pilots are often forced to temporarily use the time option during a flight to maintain flight schedules, although the use of this option increases fuel costs (Cook et al., 2009). The NO_x (Mulder and Ruijgrok, 2008) and contrail options (Fichter et al., 2005; Mannstein et al., 2005; Gierens et al., 2008; Sridhar et al., 2011; Schumann et al., 2011; Rosenow et al., 2017) have been examined as a routing strategy towards climate impact reduction. Moreover, conflicting scenarios (trade-offs) between different routing strategies have been studied; for example, Irvine et al. (2014) assessed a trade-off between climate impact of contrails and extra CO_2 emission ($\sim$ extra fuel use) for a single flight. AirTraf enables analyzing those subjects. Last but not least, the aCCFs are new proxies for the climate-optimized routing. An important aim of the AirTraf development is to verify the aCCFs themselves and the routing strategy based on the aCCFs (i.e., the climate option) in multi-annual (long-term) simulations (Yin et al., 2018b).

This paper is organized as follows. Section 2 describes an overview of AirTraf 2.0. Particularly, key changes in the model components are stated. Section 3 presents the results and discussion for the example applications of AirTraf 2.0 using the nine routing options. Section 4 verifies the consistency of the results with literature data. Finally, Sect. 5 concludes this study.

2 Overview of AirTraf 2.0

2.1 Chemistry-climate model EMAC

The EMAC model is a numerical chemistry and climate simulation system that includes submodels describing tropospheric and middle atmosphere processes and their interaction with oceans, land, and influences coming from anthropogenic emissions (Jöckel et al., 2010, 2016). It uses the second version of the Modular Earth Submodel System (MESSy2) to link multi-institutional computer codes. The core atmospheric model is the 5th generation European Centre Hamburg general circulation model (ECHAM5; Roeckner et al., 2006). For the present study, we applied EMAC (ECHAM5 version 5.3.02 and MESSy version 2.53 updated from the version 2.41 for AirTraf 1.0) in the T42L31ECMWF resolution, i.e. with a spherical truncation of T42 (corresponding to a quadratic Gaussian grid of approximately $2.8°$ by $2.8°$ in latitude and longitude) with 31 vertical hybrid pressure levels up to $10\,\mathrm{hPa}$ (middle of the uppermost layer). The namelist setup for ECHAM5 simulations (referred to the E5 setup, no chemistry) was employed. Moreover, the submodel AirTraf was coupled to the submodel CONTRAIL (version 1.0; Frömming et al., 2014) for the contrail option, and to the submodel ACCF (version 1.0) for the climate option, using the MESSy interfaces. Further information about MESSy, including the EMAC model system, is available from the MESSy Consortium Website (http://www.messy-interface.org).

2.2 Model components of submodel AirTraf

Figure 1 shows the flowchart of the submodel AirTraf 2.0. The present version is based on the model components of Air-Traf 1.0. Thus, this section outlines them. First, air traffic data and AirTraf parameters are read in the main entry point `messy_initialize` (Fig. 1, dark blue). They consist of a one-day flight plan (including departure and arrival airport

5   pairs, latitude and longitude of the airports, and departure time), Eurocontrol's Base of Aircraft Data (BADA Revision 3.9; Eurocontrol, 2011), ICAO engine performance data (ICAO, 2005), a load factor, jet fuel price, an aircraft routing option, etc. Any arbitrary number of flight plans is applicable and is reused for AirTraf simulations longer than two days. Table 1 lists the relevant data of an A330-301 aircraft and constant parameters used in AirTraf 2.0. Second, all the entries are distributed in parallel by the message passing interface (MPI) standard (called for the main entry point `messy_init_memory`; Fig.

10   1, blue). Third, the air traffic simulation (called the AirTraf integration; Fig. 1, light blue) is called in the main entry point `messy_global_end`, considering local atmospheric conditions for every flight route. The AirTraf integration uses three modules: the aircraft routing module (Fig. 1, light green), the fuel-emissions-cost-climate calculation module (Fig. 1, light orange), and the flight trajectory optimization module (Fig. 1, dark green). The first module calculates flight trajectories corresponding to a selected routing option. The second module comprises a total energy model based on the BADA methodology

15   (Eurocontrol, 2011; Schaefer, 2012) and the DLR fuel flow method (Deidewig et al., 1996). The third module consists of the Adaptive Range Multi-Objective Genetic Algorithm (ARMOGA version 1.2.0; Sasaki et al., 2002; Sasaki and Obayashi, 2004, 2005). Finally, simulation results are gathered from the MPI tasks. Optimum flight trajectories and global fields of flight properties (four-dimensional Gaussian grid; Fig. 1, rose red) are output. The same assumptions made in AirTraf 1.0 are applied in AirTraf 2.0, e.g., only the cruise flight phase is considered; trajectory conflicts and operating constraints (e.g., military air

20   space) are neglected. Further details of the model components have been reported by Yamashita et al. (2016).

2.3 Calculation procedures of the AirTraf integration

AirTraf 2.0 follows the calculation procedures of AirTraf 1.0 described in detail in Sect. 2.4 of Yamashita et al. (2016). This section reviews the procedures of the AirTraf integration (Fig. 1, light blue) with emphasis on changes by introducing the new routing options.

25   A one-day flight plan includes departure time for every flight. A flight moves to the flying process (dashed box in Fig. 1, light blue) according to individual departure time in the time loop of EMAC. The flying process comprises four steps: flight trajectory calculation, fuel-emissions-cost-climate calculation, aircraft position calculation, and gathering global emissions (bold-black boxes in Fig. 1, light blue). The first step finds an optimum flight trajectory for a selected routing option by using the aircraft routing module (Fig. 1, light green). The flight trajectory optimization module (Fig. 1, dark green) executes the

30   flight trajectory optimization under atmospheric conditions at the departure day and time of the flight. Thus, the optimum flight trajectory varies day by day. Note that the three-dimensional wind components (u, v, w) are considered in the flight trajectory optimization for all routing options. The resulting optimum flight trajectory consists of waypoints $(i = 1, 2, \cdots, n_{\mathrm{wp}})$ and flight segments $(i = 1, 2, \cdots, n_{\mathrm{wp}} - 1)$, where i is the index arranged from the departure $(i = 1)$ to the arrival $(i = n_{\mathrm{wp}})$, and n_{wp} is

the number of waypoints (see Fig. 3 of Yamashita et al., 2016). Table 2 lists flight properties calculated for the waypoints, the flight segments, and the whole trajectory. In AirTraf 2.0, 15 new properties are calculated.

The second step, which is linked to the fuel-emissions-cost-climate calculation module (Fig. 1, light orange), calculates the flight properties of fuel, NO_x emission, COC, etc. under the atmospheric conditions (Table 2, third group). This calculation is
5   performed once at the departure time of the flight. Details of the fuel-emissions-cost-climate calculation module (Fig. 1, light orange) and its reliability have been reported in Sects. 2.5, 2.6, and 5 of Yamashita et al. (2016).

The third step moves the aircraft to a new position along the optimum flight trajectory corresponding to the time steps of EMAC, by referring to the estimated time when the aircraft passes through the waypoints (called the estimated time over ETO, Table 2).

10   At the fourth step, the individual flight properties corresponding to a flight path for one time step of EMAC are gathered into the aforementioned global fields: NO_x emission, H_2O emission, fuel use, flight distance, contrail distance (PCC_{dist}), and average temperature responses for the time horizon of 20 years (ATR20s of ozone, methane, water vapor, CO_2, contrails, and total; see Sect. 2.5.7) are gathered along the flight segments (Table 2). If the aircraft reaches the last waypoint in the time loop of EMAC, the aircraft has landed (i.e., the flight quits) and the flying process ends for this flight.

15   ## 2.4   Flight trajectory optimization

The flight trajectory optimization methodologies described by Yamashita et al. (2016) are also used for the new routing options and are outlined in this section. The flight trajectory optimization module (Fig. 1, dark green) executes the optimization. The module consists of ARMOGA (version 1.2.0; Sasaki et al., 2002; Sasaki and Obayashi, 2004, 2005), which is a stochastic optimization algorithm.

20   A solution $\boldsymbol{x}$ (the term is synonymous with the flight trajectory) is a vector of n_{dv} design variables: $\boldsymbol{x} = (x_1, x_2, \cdots, x_{n_{dv}})^T$, here $n_{dv} = 11$. With the design variable index j ($j = 1, 2, \cdots, n_{dv}$), $x_j (j = 1, 2, \cdots, 6)$ indicate longitudes and latitudes, and $x_j (j = 7, 8, \cdots, 11)$ indicate altitudes. The jth design variable varies between lower and upper bounds $[x_j^l, x_j^u]$. The bounds of $[x_j^l, x_j^u]$ ($j = 1, 2, \cdots, 6$) are automatically set for a given airport pair, whereas those of $[x_j^l, x_j^u]$ ($j = 7, 8, \cdots, 11$) are set as $[x_j^l, x_j^u] = [FL290, FL410]$ (flight levels; FL290 and FL410 denote 29 000 and 41 000 ft, respectively). Geographic locations
25   of the airport pair are set according to the flight plan; altitudes of the airport pair are set to FL290. Given values of $x_j (j = 1, 2, \cdots, n_{dv})$, a three-dimensional flight trajectory is represented by a B-spline curve (third-order) between the airport pair (an illustration is given in Fig. 6 of Yamashita et al., 2016).

The initial population operator (Fig. 1, dark green) generates initial values of $x_j (j = 1, 2, \cdots, n_{dv})$ at random within the lower and upper bounds, and creates an initial population, which consists of a random set of solutions (population approach; the population size is set by n_p). An objective function f is defined, depending on a selected routing option (see Sect. 2.5), and a single-objective optimization problem can be written as follows:

$$\left. \begin{aligned} \text{Minimize} \quad & f \\ \text{Subject to} \quad & x_j^l \le x_j \le x_j^u, \quad j = 1, 2, \cdots, n_{dv} \end{aligned} \right\}, \tag{1}$$

where no constraint function is used. The ARMOGA solves the optimization problem by the following genetic operators: evaluation, selection, crossover, and mutation (Fig. 1, dark green; Holand, 1975; Goldberg, 1989). A value of f is calculated for each of the solutions by the evaluation operator. In this study, good solutions were identified in the population by the Fonseca-Fleming Pareto ranking method (Fonseca et al., 1993); the stochastic universal sampling selection (Baker, 1985) was

5 used for the selection operator; the Blend crossover operator (BLX-alpha; Eshelman, 1993) was applied to the population to create new solutions (child solutions) by picking two solutions (parent solutions) from the population; the revised polynomial mutation operator (Deb and Agrawal, 1999) was used to add a disturbance to the child solutions. When the evolution process is iterated for a number of generations (this is set by n_g), the population of solutions is improved by reducing f, and another superior population is created in subsequent generations. Finally, the ARMOGA finds the best solution (one optimum flight

10 trajectory) with the minimum value of f through the whole generations; the flight properties of the solution are stored, as shown in Table 2. The flight trajectory optimization stated above is executed for every airport pair. Detailed descriptions of the optimization methodologies, appropriate ARMOGA parameter settings, and the accuracy of the optimization module have been presented in Sect. 3.2 of Yamashita et al. (2016).

2.5 Formulations of objective functions for new aircraft routing options

15 Seven new objective functions were developed for the new aircraft routing options. The following subsections describe formulations of the objective function f for those options. To calculate f, the fuel-emissions-cost-climate calculation module (Fig. 1, light orange) is used as necessary by the evaluation operator (Fig. 1, dark green) in the flight trajectory optimization.

2.5.1 Fuel use

The objective function for the fuel option represents the sum of fuel use kg(fuel) of a flight:

20 $$f = \sum_{i=1}^{n_{\mathrm{wp}}-1} \mathrm{FUEL}_i, \tag{2}$$

where FUEL_i is the fuel use of the ith flight segment (Table 2).

2.5.2 NO_x emission

The objective function for the NO_x option represents the sum of NO_x emission g(NO_x) of a flight:

$$f = \sum_{i=1}^{n_{\mathrm{wp}}-1} \mathrm{NO}_{x,i} = \sum_{i=1}^{n_{\mathrm{wp}}-1} (\mathrm{FUEL}_i \mathrm{EINO}_{x,a,i}), \tag{3}$$

25 where $\mathrm{NO}_{x,i}$ is the NO_x emission of the ith flight segment; $\mathrm{EINO}_{x,a,i}$ is the NO_x emission index under actual flight conditions at the ith waypoint (Table 2) and is calculated using the ICAO engine performance data (ICAO, 2005; see Sect. 2.6 of Yamashita et al., 2016).

2.5.3 H$_2$O emission

The objective function for the H$_2$O option represents the sum of H$_2$O emission g(H$_2$O) of a flight:

$$f = \sum_{i=1}^{n_{wp}-1} \mathrm{H_2O}_i = \mathrm{EIH_2O} \sum_{i=1}^{n_{wp}-1} \mathrm{FUEL}_i, \tag{4}$$

where H$_2$O$_i$ is the H$_2$O emission of the ith flight segment (Table 2); EIH$_2$O is the emission index of H$_2$O and was set as EIH$_2$O
5   = 1230 g(H$_2$O)(kg(fuel))$^{-1}$ (Table 1). The H$_2$O emission is proportional to the fuel use by assuming an ideal combustion of jet fuel. Thus, this option yields the same results as the fuel option in AirTraf 2.0. If an alternative fuel option is introduced, the H$_2$O option probably differs from the fuel option, because the emission index may not be constant.

2.5.4 Contrail formation

Yin et al. (2018a) and Yin et al. (manuscript in preparation, 2019) developed the routing option for contrail avoidance. This
10   option avoids regions where persistent contrail formation is expected. The objective function represents a total contrail distance km(contrail) of a flight:

$$f = \sum_{i=1}^{n_{wp}-1} \mathrm{PCC}_{\mathrm{dist},i} = 10^{-3} \sum_{i=1}^{n_{wp}-1} (\mathrm{Potcov}_i d_i), \tag{5}$$

where PCC$_{\mathrm{dist},i}$ is the contrail distance of the ith flight segment; Potcov$_i$ is the potential persistent contrail cirrus coverage at the ith waypoint; and d_i is the flight distance of the ith flight segment (Table 2). PCC$_{\mathrm{dist},i}$ is calculated by using the potential
15   contrail coverage Potcov (Ponater et al., 2002; Burkhardt et al., 2008; Burkhardt and Kärcher, 2009; details of Potcov have been reported by Frömming et al., 2014). The Potcov represents fractional areas in which contrails can maximally occur under a given atmospheric condition. The Potcov is calculated by the submodel CONTRAIL (version 1.0; Frömming et al., 2014), using a parameterization scheme based on the thermodynamic theory of contrails, i.e., the Schmidt-Appleman theory (Schmidt, 1941; Appleman, 1953; Schumann, 1996).

20   ### 2.5.5 Simple operating cost (SOC)

The cost index (CI) is set during a real flight to manage airline operation costs and is defined as the ratio of time cost to fuel cost (CI = $C_{\mathrm{time}}/C_{\mathrm{fuel}}$, where C denotes a cost). A low CI value causes an aircraft to minimize fuel use with a sacrifice of flight time, which enables a long-range flight. Conversely, a high CI value causes the aircraft to minimize flight time with an extra fuel use. Generally, the operating costs are a function of flight time and fuel. Thus, the minimum cost solution lies in a
25   trade-off between flight time and fuel (Cook et al., 2009; Marla et al., 2016). Here, the objective function simply represents the sum of the time and the fuel costs on the basis of the CI features:

$$f = \mathrm{SOC} = c_t \sum_{i=1}^{n_{wp}-1} \frac{d_i}{V_{\mathrm{ground},i}} + c_f \sum_{i=1}^{n_{wp}-1} \mathrm{FUEL}_i, \tag{6}$$

where c_t and c_f are the unit costs of time and fuel, respectively (Table 1); $V_{\mathrm{ground},i}$ is the ground speed at the ith waypoint (Table 2). Note that, the c_t includes the cost elements for flight crew, cabin crew, and maintenances for both, airframe and engines.

2.5.6 Cash operating cost (COC)

5 The COC is a comprehensive economic criterion for evaluating airline operation costs (Liebeck et al., 1995). The COC includes the cost elements for flight crew, cabin crew, landing fee, navigation fee, fuel, and maintenances for both, airframe and engines (no costs for depreciation, insurance, and interest are included). The COC calculation method for international flights (Liebeck et al., 1995) was employed. Those cost elements were calculated on the basis of the price in 1993 and were scaled to 2015 by the average U.S. inflation rate of average consumer prices r_{inf} (Table 1; IMF, 2016). Only the fuel cost was

10 directly calculated with the current jet fuel price JFP (Table 1; IATA, 2017). A block time and a block fuel originally used in the method were replaced by the total flight time FT and the fuel use of $\sum_{i=1}^{n_{\mathrm{wp}}-1} \mathrm{FUEL}_i$ in AirTraf 2.0, respectively (Table 2). The objective function can be written as:

$$f = \mathrm{COC} = C_{\mathrm{flightcrew}} + C_{\mathrm{cabincrew}} + C_{\mathrm{landing}} + C_{\mathrm{navigation}} + C_{\mathrm{fuel}} + C_{\mathrm{airframe}} + C_{\mathrm{engine}}. \tag{7}$$

A detailed description of the COC calculation method has been reported in Liebeck et al. (1995). Given the parameters and

15 variables listed in Tables 1 and 2, Eq. (7) becomes a function of the flight time and the fuel.

2.5.7 Climate impact

The climate-optimized routing was carried out by using the aCCFs (Van Manen, 2017; Yin et al., 2018a; Van Manen and Grewe, 2019; Yin et al. (manuscript in preparation, 2019)) calculated by the submodel ACCF. The aCCFs are approximation functions based on regression analyses for the simulated CCFs data (see Sect. 1); the aCCFs represent a correlation of meteorological

20 variables at the time of flight with anticipated climate impacts, i.e., ATR20s of ozone, methane, water vapour, CO_2, and contrails are estimated on a per unit basis (the definition of the aCCFs are given in the Appendix and examples are shown in Fig. S1 in the Supplementary material). These five ATR20s are calculated for flight segments (Table 2) and are combined into an objective function to represent an anticipated climate impact of a flight (in K):

$$\mathrm{ATR20}_{\mathrm{total},i} = \mathrm{ATR20}_{\mathrm{O}_3,i} + \mathrm{ATR20}_{\mathrm{CH}_4,i} + \mathrm{ATR20}_{\mathrm{H}_2\mathrm{O},i} + \mathrm{ATR20}_{\mathrm{CO}_2,i} + \mathrm{ATR20}_{\mathrm{contrail},i}, \tag{8}$$

25 $$f = \sum_{i=1}^{n_{\mathrm{wp}}-1} \mathrm{ATR20}_{\mathrm{total},i}, \tag{9}$$

where $\mathrm{ATR20}_{\mathrm{contrail},i}$ can take positive and negative values, because of the day-time and night-time contrail effects. The aCCFs are derived based on the CCFs data of the north-Atlantic region and are applicable to the northern and high latitudes. Further details of the aCCFs have been reported in the literature mentioned above.

3 Example application: one-day simulation with new aircraft routing options

3.1 Simulation setup

Nine one-day simulations were carried out for a demonstration of AirTraf 2.0. Table 3 lists the simulation setups. The same setups that we used for the consistency check for AirTraf 1.0 simulations (Yamashita et al., 2016) were employed; only the
5 simulation period was changed into a recent day, which showed a typical weather condition in winter with a strong jet stream (see Fig. S2 in the Supplementary material). The flight altitude for the great circle option was set to FL350; the altitude for the other options was calculated in the trajectory optimization within [FL290, FL410], as mentioned in Sect. 2.4. The trans-Atlantic flight plan (103 flights) of an Airbus A330 aircraft was provided by Grewe et al. (2014a) and REACT4C (2014). The setups for the optimization parameters were determined by the benchmark tests (Yamashita et al., 2016).

10 ### 3.2 Optimized flight trajectories and global fields

To verify the simulation output, the obtained optimized trajectories and global fields for the contrail, the COC, and the climate options are shown. Figure 2 shows the optimized trajectories for those options (optimized trajectories for other options are shown in Supplement Fig. S3). Obviously, the optimum trajectories vary with the routing options. Figures 2c and 2d show that the COC optimum trajectories of the eastbound flights leap up over the North Atlantic Ocean, whereas these trajectories
15 are shifted northward for the westbound flights. As the jet stream is located at around 50°W and 40°N (see Fig. S2 in the Supplementary material), the eastbound trajectories are optimized to benefit from tailwinds of the jet stream and the westbound trajectories avoid headwinds of the jet by detouring northward. In addition, most of those trajectories are located at high flight altitudes (∼FL410, 12.5 km), because fuel consumption decreases due to aerodynamic drag reduction (Fichter et al., 2005; Schumann et al., 2011; Yamashita et al., 2016). In contrast, the contrail and the climate options show complex shaped
20 trajectories with various flight altitude changes (see Figs. 2a, 2b, 2e, and 2f).

Figure 3 shows the global fields of fuel use, contrail distance, and climate impact indicated by $ATR20_{total}$ for the three options. We see that the contrail option certainly decreases the contrail distance; the COC option shows the narrower fuel distribution than that of the contrail and climate options; and the climate option shows the net climate impact reduction (the local negative values, i.e. cooling effects, are mainly caused by contrails). A comprehesive analysis of the optimized trajectories
25 for the calculated fields is beyond the scope of this paper. However, it is apparent from Fig. 3 that the optimized trajectories successfully decrease the respective objects (target measures) to be minimized (this point is discussed quantitatively in Sect. 3.3).

3.3 Characteristics of aircraft routing options

To examine the characteristics of the routing options, Table 4 lists a summary of typical performance measures of the one-day
30 air traffic (total 103 flights) for specific routing options (bar charts are given in Supplement Fig. S4). Relative changes (in %) to the COC option are also listed in Table 4, considering this option as a reference (the COC option is assumed to be the current

aircraft routing strategy). Table 4 shows that individual options successfully minimize their own object (target measure; see measures marked with an asterisk in Table 4). These results confirm the correctness of the new routing options in AirTraf, since we solve a single-objective minimization problem defined by Eq. (1) for each routing option.

We see from Table 4 that the great circle option has the minimum flight distance, whereas this option increases other mea-

5 sures. The time option shows the minimum flight time with a penalty on fuel use, NO_x emission, H_2O emission, SOC, COC, and $ATR20_{total}$ (further discussion in Sect. 4). The fuel (and also the H_2O), the NO_x, the SOC, and the COC options obtain similar values on all the measures (see also Supplement Fig. S4). Of the nine routing options, these options show decreased fuel use, NO_x and H_2O emissions, SOC, and COC, whereas contrail distance and $ATR20_{total}$ increase. The difference among these options is considered significant for airline operations and thus is discussed in Sect. 4. The contrail option shows the minimum

10 contrail distance and decreases $ATR20_{total}$, whereas the other measures considerably increase. This option allows aircraft to widely detour the potential contrail regions (because no constraint function is used in Eqs. (1) and (5)). Thus, the flight time and the fuel use drastically increase, which results in the increase of NO_x and H_2O emissions, SOC, and COC. The contrail option shows the highest SOC and COC of the nine routing options. Comparing the contrail option with the COC option indicates that the contrail distance decreases with an additional fuel use of 8.3 $kg(fuel)(km(contrail))^{-1}$ (i.e., the additional COC of

15 6.20 $USD(km(contrail))^{-1}$). The SOC and the COC options are comparable. The two options show similar values for all the measures and have the same minimum SOC and COC. This is because the objective function of the two options is a function of flight time and fuel. In fact, the obtained optimum trajectories for the SOC and the COC options are approximately the same (see Figs. 2c, 2d and Supplement Figs. S3k and S3l). The climate option achieves the minimum $ATR20_{total}$ and decreases the contrail distance, whereas this option increases all other measures, particularly SOC and COC increase sharply. The present

20 results indicate that the contrail and the climate options considerably reduce the climate impact indicated by $ATR20_{total}$. The cost-benefit performance (i.e., the COC increment per $ATR20_{total}$ reduction) for the contrail and the climate options are 0.24 and 0.13 $Mil.USD(10^{-7}K)^{-1}$, respectively. Thus, the climate option seems to be a more cost-effective option. Note that this performance is a narrow result obtained using AirTraf 2.0 under the specific conditions (e.g., the simulations were carried out with the 103 north-Atlantic flights on December 1, 2015, as shown in Table 3).

25 4   Discussion: verification of the one-day AirTraf simulation results

This paper presents the extended version of the submodel AirTraf, which offers additional aircraft routing options for defining overall target functions for aircraft trajectory optimization. To confirm the consistency of AirTraf simulations, the relative changes in the performance measures among the routing options (listed in Table 4 in parentheses) are compared with previous studies. The quantitative values of the changes in performance measures vary, depending on different methodologies,

30 atmospheric conditions, simulation periods, flight plans, aircraft/engine types, cost/climate impact metrics, etc. Thus, a direct comparison in magnitude of our results with published studies is difficult; the sign of the relative changes in the measures is compared. Note that the great circle and the time options have been verified before (Yamashita et al., 2016). In addition, the

H_2O option yields the same results as the fuel option (see Sect. 2.5.3 and Table 4); the SOC option is comparable to the COC option (see Sect. 3.3 and Table 4). Thus, we omit any discussion of the H_2O and the SOC options here.

First, the time, the fuel, and the COC options are analyzed. As defined in Sect. 2.5.6, COC is a combined function of flight time and fuel. To minimize COC, one may attempt to reduce both factors simultaneously; however, a trade-off between the

5 flight time and the fuel generally exists. Table 4 shows that the time penalty of flying minimum fuel trajectories is 2.4 percentage points (%pt), whereas the fuel penalty of flying minimum time trajectories is 20.3 %pt. A similar trade-off was reported by two published studies. Celis et al. (2014) addressed a single-objective flight trajectory optimization on total flight time and fuel use, respectively, under International Standard Atmospheric (ISA) conditions. A typical single-aisle aircraft (150 passengers) with twin turbofan engines was assumed; the aircraft speed and the flight altitude in eight flight segments were optimized for

10 a given flight trajectory (a quasi-full flight profile optimization). Compared to the minimum time trajectory, the fuel optimum trajectory decreased the fuel use by 31.7 %pt with increasing flight time by 14.0 %pt. Rosenow and Fricke (2016) compared performances for the minimum time and the minimum fuel trajectories for a flight from Frankfurt (Main) to Dubai for a Boeing B777 freighter on February 2, 2016, at 12 a.m. The comparison showed that the fuel optimum trajectory decreased fuel use by 8.0 % with increasing time by 3.7 %. These studies imply that the minimum COC solution lies between the minimum time

15 and the minimum fuel solutions. In fact, Table 4 shows that the COC option has more flight time than that of the time option, and that the COC option consumes more fuel than that of the fuel option. The COC option yields the values of compromise (i.e. not minimum) of flight time and fuel. Nonetheless, this option achieves the minimum COC. The submodel AirTraf 2.0 can consistently differentiate those three solutions.

To support the discussion above, the fuel and the COC options are compared in detail. Erzberger and Lee (1980) compared

20 the minimum fuel and the minimum direct operating cost (DOC) trajectories for a short-haul route for a Boeing 727-100 aircraft on the basis of optimum control theory (Bryson and Ho, 1969) under U.S. Standard Atmospheric conditions. They showed that flying "minimum fuel" reduced fuel use by 6.9 %, whereas the time and the DOC penalties of the trajectory were 23 and 6 %, respectively (constrained thrust case). Our results in Table 4 show that the fuel option reduces fuel use by 0.1 %, whereas the time and the COC penalties of the option are 0.1 and 0.03 %, compared to those measures of the COC option. The signs of

25 these relative changes obtained from our results agree with those shown by Erzberger and Lee (1980). In addition, the time and the COC options are compared in a perspective of airline operating economics. Although the time option increases fuel use, NO_x emission, H_2O emission, SOC, COC, and $ATR20_{total}$ (fuel use and COC increase by 20.2 and by 6.1 %, respectively), the option decreases flight time by 2.3 %, compared to that of the COC option. In other words, the time option reduces flight time with the extra cost of 19 034.74 $USDh^{-1}$ (= 269.66 $EURmin^{-1}$; converted by 1 USD = 0.85 EUR on September 18,

30 2018 (European Central Bank, 2018)). In a context of delay recovery, this extra cost is the same order of magnitude to flight delay costs. If the flight delay costs exceed the extra cost due to the time option, operators would determine to fly faster by using the time option to recover the delay. Cook et al. (2004, 2009) reported that the flight delay costs, which are associated with delayed passengers, additional fuel use, flight crew, cabin crew, and marginal maintenance costs, reached several hundred Euros per minute. The extra cost calculated from our results agrees well with this report.

Compared to the COC option, the NO_x option decreases the NO_x emission by 0.5 %, leading to a COC increase of 0.2 %. Mulder and Ruijgrok (2008) analyzed effects of varying cruise conditions on NO_x emission and on DOC from the cruise NO_x simulation model (Bremmers, 1999) by assuming a cruise range of 5800 km with a Boeing 747-400 aircraft under ISA conditions. They clearly concluded that a reduction of NO_x emission caused a cost increase. Our results agree well with this

5 conclusion. Moreover, the NO_x option differs from the fuel option, because the amount of NO_x emission depends not only on fuel use, but also on the NO_x emission index, as defined in Eq. (3). The emission index depends strongly on the ambient atmospheric conditions at every waypoint (see Sect. 2.6 of Yamashita et al., 2016). Table 4 shows that the NO_x option decreases the NO_x emission by 0.3 %pt, whereas this option increases flight time by 0.2 %pt and fuel use by 0.2 %pt, compared to those measures of the fuel option. Celis et al. (2014) addressed a single-objective flight trajectory optimization on total fuel use and

10 on NO_x emission, respectively, with the same simulation setup described above. Compared to the minimum fuel trajectory, the minimum NO_x trajectory decreased NO_x emission by 10.4 %pt, whereas the trajectory increased time by 1.0 %pt and fuel use by 3.9 %pt. The signs of the relative changes obtained from our results are in good agreement with those shown by Celis et al. (2014).

The contrail option drastically decreases contrail distance by 79.8 % and $ATR20_{total}$ by 43.4 %, whereas this option in-
15 creases fuel use by 23.0 % and COC by 12.0 %, compared to those measures of the COC option. The contrail option is effective in order to reduce the climate impact, as pointed out by previous studies introduced in Sect. 1. Here, those relative changes in the measures are compared with two published studies. Rosenow et al. (2017) performed a one-day European's air traffic optimization on July 25, 2016. The total number of 13 584 flights over Europe (containing 16 aircraft types) was employed; their three dimensional flight profiles were optimized for airline costs (termed as the cost performance indicators CPI) and en-
20 vironmental impacts (termed as the ecological performance indicators EPI). They revealed that an additional contrail avoidance intent decreased contrail costs by 31.5 % (contrail formations were converted into a monetary value) and EPI by 5.2 %, whereas the intent increased fuel use by 0.05 % and CPI by 0.5 % over those of the minimum cost strategy. The signs of the relative changes obtained from our simulations are consistent with those shown by Rosenow et al. (2017). Furthermore, Sridhar et al. (2013) applied a contrail reducing strategy to aircraft flying between 12 airport pairs (287 flights) in the United States on April
25 12, 2010. The three-dimensional contrail reducing strategy showed a trade-off between contrail formation time (time spent in traveling through contrail formation regions) and fuel consumption. Representative points on the trade-off curve showed that the contrail formation time decreased by 4415 and by 5301 min with an additional fuel use of 20 000 and of 131 000 kg(fuel), respectively, over those of a wind-optimal strategy (this strategy is regarded as an economically optimal strategy; see Sect. 2.4 of Yamashita et al., 2016). This study clearly indicated the fuel increase by avoiding contrail formations. Our results agree well
30 with the finding of Sridhar et al. (2013).

Table 4 clearly shows a trade-off between economic cost and climate impact (see also Supplement Fig. S4). Compared to the COC option, the climate option decreases $ATR20_{total}$ by 67.9 % with an additional COC of 9.8 %. A similar trade-off certainly exists between the minimum COC and the minimum climate impact trajectories for each airport pair. The trade-off obtained from our results agrees with that indicated by many studies (see Sect. 1). Moreover, Niklaß et al. (2017) performed an
35 aircraft trajectory optimization for nine north-Atlantic flight routes varying weighting factors on average temperature response

over 100 years (ATR100) and on COC under ISA conditions. They showed a clear trade-off between the cost and the climate impact. The minimum climate impact trajectories, on average, reduced ATR100 by 28.4 % with an additional COC of 7.1 %, compared to those measures of the minimum COC trajectories. Our results agree with those shown by Niklaß et al. (2017).

As discussed above, the many previous studies verify the consistency of the AirTraf simulations. Before concluding the
5 discussion, two superior aspects of the AirTraf submodel are emphasized, compared to the simulation models used in the previous studies. First, AirTraf enables an intercomparison for various aircraft routing options all at once, because all the options are integrated. Normally, one or two specific routing options are available for a flight trajectory optimization in other models. Second, AirTraf performs air traffic simulations not under ISA conditions, not under a fixed atmospheric condition for a specific day, but under comprehensive atmospheric conditions which are calculated by the chemistry-climate model
10 EMAC. AirTraf can simulate air traffic for long-term period in EMAC, which enables one to examine effects of aircraft routing strategies on climate impact on a long time scale.

5   Conclusions

We introduced updates to the air traffic simulation model AirTraf in the chemistry-climate model EMAC. The submodel AirTraf 2.0 was developed according to the MESSy standard and was described in detail in this paper. This submodel introduces seven
15 new aircraft routing options for air traffic simulations: the fuel use, the NO_x emission, the H_2O emission, the contrail formation, the simple operating cost, the cash operating cost, and the climate impact options. Our flight trajectory optimization methodology consists of genetic algorithms; the methodology was similarly used and was validated beforehand (Yamashita et al., 2016). The particular strength of AirTraf is to enable a flight trajectory optimization for a global flight movement set in the atmosphere which is comprehensively described by EMAC. The novel routing option, i.e., the climate impact option, has been integrated
20 in AirTraf 2.0. This option uses meteorological variables in terms of (spatially and temporally varying) aviation climate impact estimated by the aCCFs, and optimizes flight trajectories by minimizing their anticipated climate impact. As the aCCFs are new proxies for the climate-optimized routing, AirTraf takes a role in verifying the aCCFs themselves and the climate impact option based on the aCCFs in multi-annual (long-term) simulations.

To verify the submodel AirTraf 2.0, example simulations were carried out with 103 north-Atlantic flights of an Airbus A330
25 aircraft for a typical winter day. The AirTraf 2.0 simulates the one-day air traffic successfully for each routing option. The characteristics of the routing options were analyzed on the basis of the nine performance measures. Aircraft was flown as the minimum economic cost with both, the SOC and the COC options. These options are comparably effective for economic cost indices. The AirTraf 2.0 differentiates the minimum time, the minimum fuel, and the minimum COC solutions; that is, the COC option obtains a trade-off solution between the minimum time and the minimum fuel solutions. The NO_x option minimizes
30 NO_x emission; this option differs from the fuel and the COC options. The contrail and the climate options decrease climate impact (indicated by $ATR20_{total}$), which causes extra operating costs. A trade-off between the cost and the climate impact exists. Compared to the COC option, the climate and the contrail options decrease $ATR20_{total}$ by 67.9 and by 43.4 % with an increase of COC by 9.8 and by 12.0 %, respectively. Thus, the climate option seems to be more effective on the cost-benefit

performance than the contrail option. The simulation results were compared with literature data. The relative changes in the performance measures among the various routing options agree well in sign with those shown by many previous studies. This comparison has limitations because of different methodologies, different atmospheric conditions, etc. Nonetheless, the many literature data offer evidence to indicate the consistency of the AirTraf simulations.

5   The integration of AirTraf into EMAC allows one to optimize and to study flight trajectories under historical, present-day and future conditions of the climate system. Furthermore, the integrated aircraft routing options could be extended to conflicting scenarios. Recently, Yin et al. (2018a) investigated a trade-off between flight time and contrail formation for trans-Atlantic flights, by combining the time and the contrail options. Another option could easily be created by adding a corresponding objective function. The AirTraf development presented in this paper leads to a further detailed understanding of characteristics

10   of various aircraft routing strategies.

Appendix A: The algorithmic Climate Change Functions

The aCCFs are calculated by the submodel ACCF (version 1.0). The derivation and validation of the aCCFs of ozone, methane, water vapour have been published by Van Manen (2017) and Van Manen and Grewe (2019); the aCCF of contrails is described by Yin et al. (2018a) and Yin et al. (manuscript in preparation, 2019). The aCCFs for ozone, methane, water vapour, CO_2 and

15   contrails are formulated as follows:

$$aCCF_{O_3} = \begin{cases} -5.20 \times 10^{-11} + 2.30 \times 10^{-13}T + 4.85 \times 10^{-16}\Phi - 2.04 \times 10^{-18}T\Phi, & \text{if } aCCF_{O_3} > 0, \\ 0, & \text{if } aCCF_{O_3} \leq 0, \end{cases} \tag{A1}$$

$$aCCF_{CH_4} = \begin{cases} -9.83 \times 10^{-13} + 1.99 \times 10^{-18}\Phi - 6.32 \times 10^{-16}F_{in} + 6.12 \times 10^{-21}\Phi F_{in}, & \text{if } aCCF_{CH_4} \leq 0, \\ 0, & \text{if } aCCF_{CH_4} > 0, \end{cases} \tag{A2}$$

20   $$aCCF_{H_2O} = 4.05 \times 10^{-16} + 1.48 \times 10^{-16}|PV|, \tag{A3}$$

$$aCCF_{CO_2} = 6.35 \times 10^{-15}, \tag{A4}$$

$$aCCF_{contrail} = \begin{cases} 1.0 \times 10^{-10}(0.0073 \times (10^{0.0107T} - 1.03)) \times 0.114, & \text{if } Potcov \geq 0 \ .and. \ \text{nighttime}, \\ 1.0 \times 10^{-10}(-1.7 - 0.0088 OLR) \times 0.114, & \text{if } Potcov \geq 0 \ .and. \ \text{daytime}, \\ 0, & \text{if } Potcov < 0, \end{cases} \tag{A5}$$

25   where T is the atmospheric temperature in K, Φ is the geopotential in $m^2 s^{-2}$, F_{in} is the incoming solar radiation at the top of atmosphere in Wm^{-2}, PV is the potential vorticity in PVU (1 PVU $= 10^{-6} Km^2 kg^{-1}s^{-1}$), and OLR is the outgoing longwave radiation in Wm^{-2}. Given values of these meteorological variables, Eqs. (A1) and (A2) yield $aCCF_{O_3}$ and $aCCF_{CH_4}$

in $K(kg(NO_2))^{-1}$; Eqs. (A3) and (A4) yield $aCCF_{H_2O}$ and $aCCF_{CO_2}$ in $K(kg(fuel))^{-1}$; and Eq. (A5) yields $aCCF_{contrail}$ in $K(km(contrail))^{-1}$. The $aCCF_{CO_2}$ is the sole constant value (Dahlmann, 2018). The $aCCF_{CO_2}$ is calculated by using the non-linear climate-chemistry response model AirClim (Grewe and Stenke, 2008; Dahlmann, 2012; Dahlmann et al., 2016), assuming a 1 Tg fuel use in 2010 with the annual growth rate according to the future global aircraft scenario Fa1 (Penner et al.,

5 1999). The $aCCF_{CO_2}$ is the averaged temperature response of CO_2 for the period $2010-2029$ (in K per kilogram of fuel) calculated by AirClim. In AirTraf 2.0, those five aCCFs are calculated as flight properties for waypoints and then the corresponding ATR20s are calculated for flight segments (see Table 2).

Code and data availability. AirTraf is implemented as a submodel of the Modular Earth Submodel System (MESSy). MESSy is continuously further developed and applied by a consortium of institutions. The usage of MESSy and access to the source code is licenced to all

10 affiliates of institutions which are members of the MESSy Consortium. Institutions can become a member of the MESSy Consortium by signing the MESSy Memorandum of Understanding. More information can be found on the MESSy Consortium Website (http://www.messy-interface.org). The submodel AirTraf 2.0 presented here has been developed on the basis of MESSy version 2.53 and is available since the official release of MESSy version 2.54. The status information for AirTraf including the license conditions is available on the website. The data from the simulations will be provided by the authors on request.

15 *Author contributions.* HY, FY and VG designed the submodel AirTraf 2.0. HY, FY, PJ, SM, and BK implemented the coupling of AirTraf 2.0 with the Modular Earth Submodel System (MESSy). FY, VG, SM, KD, and CF developed the algorithmic Climate Change Functions (aCCFs). FY and VG designed the submodel ACCF. HY performed the simulations and analyzed the results presented in this paper.

Competing interests. The authors declare that they have no conflict of interest.

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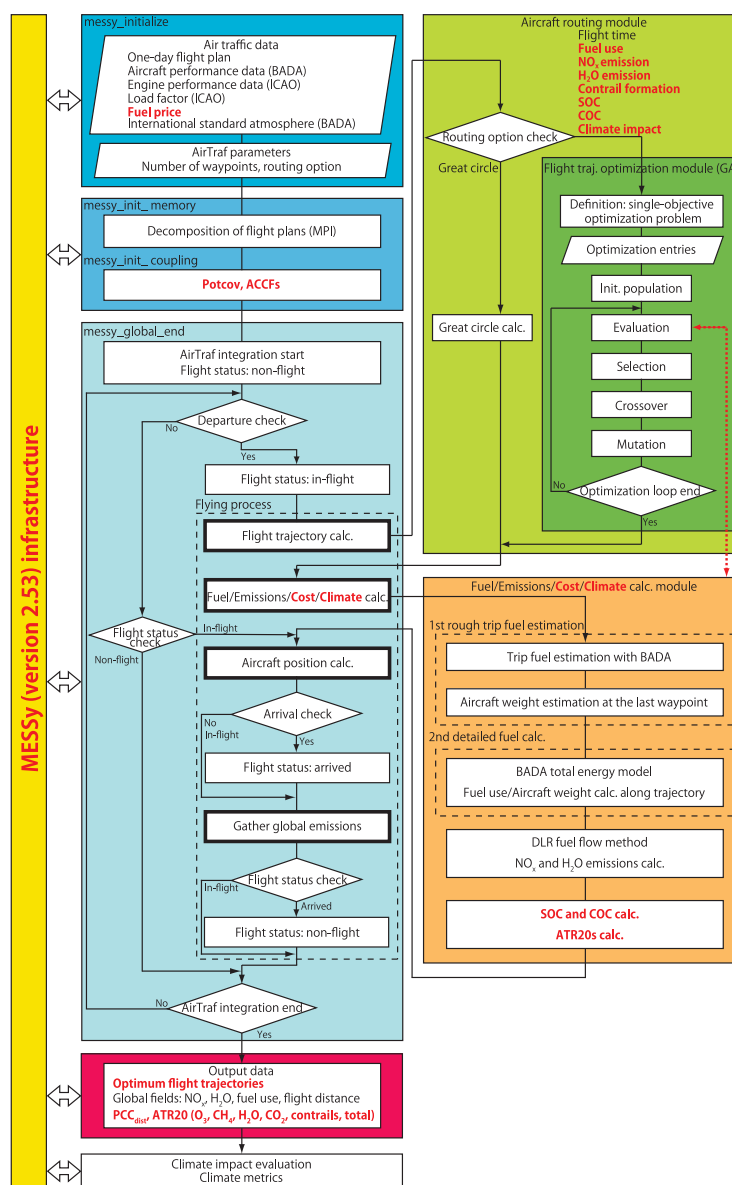
Figure 1. Updated flowchart of the MESSy submodel AirTraf 2.0 (updates from AirTraf 1.0 are highlighted by red texts and arrows). MESSy as part of EMAC provides interfaces (yellow) to couple various submodels for data exchange, run control and data input/output. AirTraf 2.0 is coupled to the submodel CONTRAIL (version 1.0; Frömming et al., 2014) and the submodel ACCF (version 1.0). Air traffic data and AirTraf parameters are imported in the initialization phase (`messy_initialize`, dark blue). AirTraf includes the flying process in `messy_global_end` (dashed box, light blue), which comprises four main computation procedures (bold-black boxes). AirTraf uses three modules: the aircraft routing module (light green), the fuel-emissions-cost-climate calculation module (light orange), and the flight trajectory optimization module (dark green). Resulting optimum flight trajectories and global fields of flight properties are output (rose red).
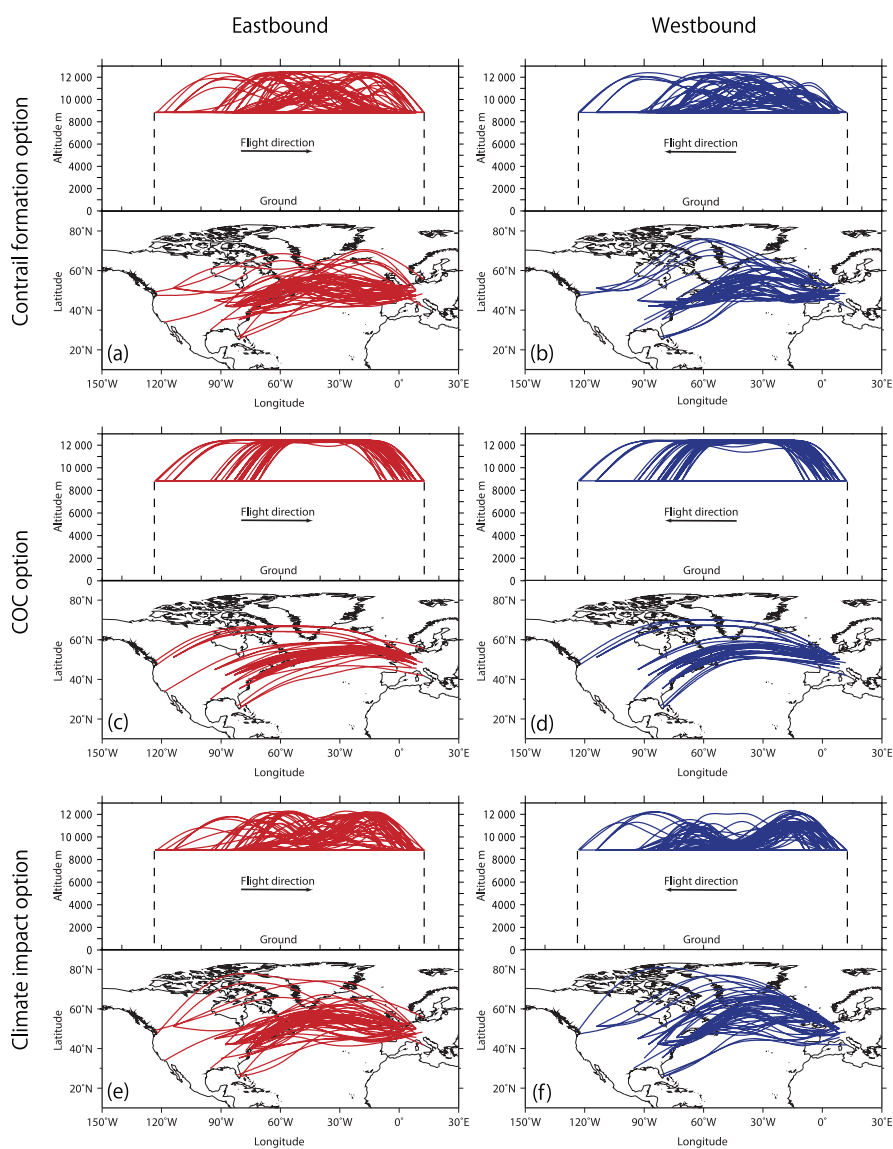
Figure 2. Optimized flight trajectories from a one-day AirTraf simulation (52 eastbound and 51 westbound flights) for the contrail formation (a, b), the COC (c, d), and the climate impact routing options (e, f). For each figure, the trajectories are shown in the vertical cross-section (top) and projected on the ground (bottom).
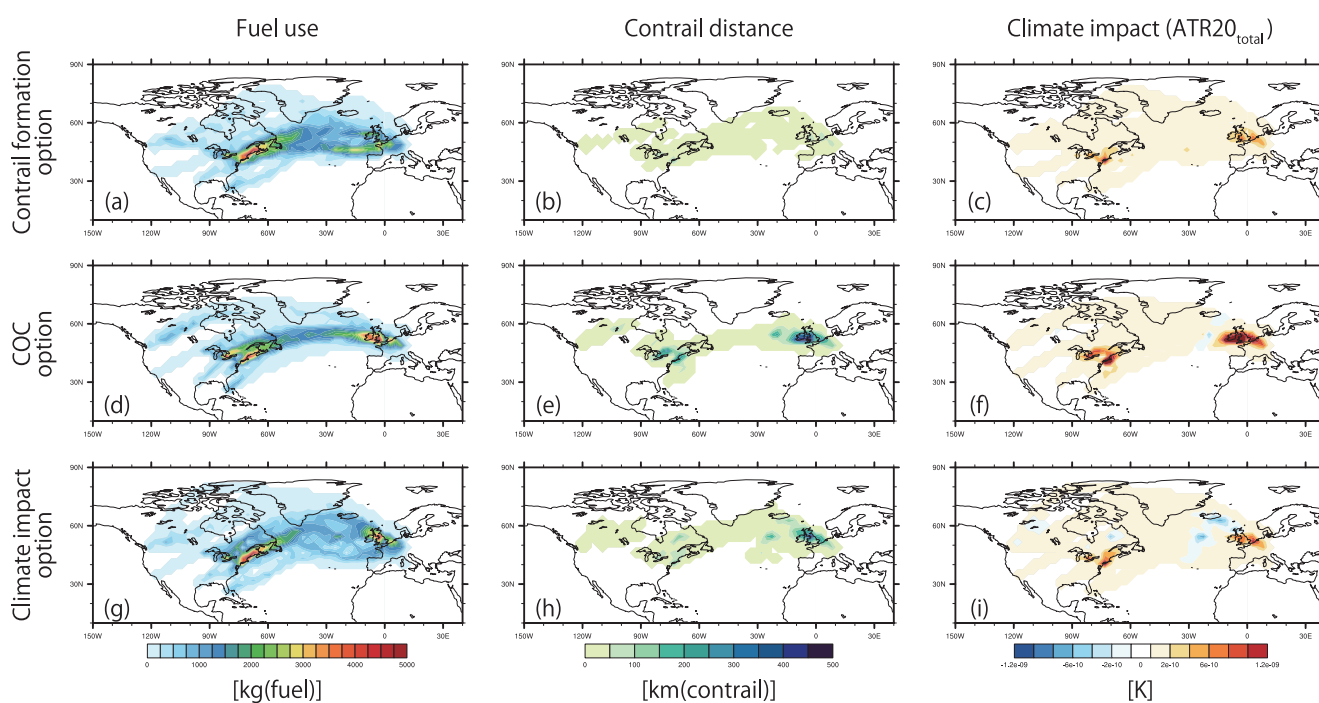
Figure 3. Vertically integrated distribution of fuel use, contrail distance, and climate impact indicated by $\mathrm{ATR20_{total}}$ for one-day (from December 1, 2015 00:00:00 to December 2, 2015 00:00:00 UTC). Top: contrail formation option. Middle: COC option. Bottom: climate impact option. These distributions were obtained with the optimized flight trajectories shown in Fig. 2 (sum of 103 flights).

Table 1. Relevant data of an Airbus A330-301 aircraft and constant parameters applied for AirTraf 2.0. The column "New in V2.0" denotes parameters newly introduced in AirTraf 2.0.

Parameter	Value	Unit	New in V2.0	Description
AFW	103 070	kg	x	Airframe weight estimated by $AFW = MEW - N_{eng}EDW$
c_t	0.75	$(USDollar)s^{-1}$	x	Unit time costs[a]
c_f	0.51	$(USDollar)kg^{-1}$	x	Unit fuel costs[a]
C_{D0}	0.019805	−		Parasitic drag coef. (cruise)[b]
C_{D2}	0.031875	−		Induced drag coef. (cruise)[b]
C_{f1}	0.61503	$kg\,min^{-1}kN^{-1}$		First thrust specific fuel consumption (TSFC) coef. (jet engines)[b]
C_{f2}	919.03	kt		Second TSFC coef.[b]
C_{fcr}	0.93655	−		Cruise fuel flow correction coef.[b]
EDW	5091.62	kg	x	Engine dry weight. CF6-80E1A2 engine[c]
$EINO_{x,ref}$	4.88; 12.66; 22.01; 28.72	$g(NO_x)(kg(fuel))^{-1}$		Reference NO_x emission index at take off, climb out, approach and idle conditions (sea level). CF6-80E1A2 (2GE051)[d]
EIH_2O	1230	$g(H_2O)(kg(fuel))^{-1}$		H_2O emission index[e]
f_{ref}	0.228; 0.724; 2.245; 2.767	$kg(fuel)s^{-1}$		Reference fuel flow at take off, climb out, approach and idle conditions (sea level). CF6-80E1A2 (2GE051)[d]
g	9.8	ms^{-2}		Gravity acceleration
JFD	0.804	kgl^{-1}	x	Jet fuel density at $15°C$ (Jet A-1)
JFP	0.41	$(USDollar)l^{-1}$	x	Jet fuel price[f]
M	0.82	−		Cruise Mach number[b]
MEW	113 253	kg	x	Baseline manufactures empty weight. $MEW = 0.9053OEW$[g]
MPL	47 900	kg		Maximum payload[b]
MTOGW	212 000	kg	x	Maximum take-off weight[h]
N_{seat}	295	−	x	Number of seats (3-class)[i]
N_{eng}	2	−	x	Number of engines[h]
OEW	125 100	kg		Operational empty weight[b]
OLF	0.62	−		ICAO overall (passenger/freight/mail) weight load factor in 2008[j]
P_0	101 325	Pa		Reference pressure (sea level)
r_{inf}	2.28	%	x	Ave. U.S. inflation rate (1994-2014)[k]
R	287.05	$JK^{-1}kg^{-1}$		Gas constant for dry air
S	361.6	m^2		Reference wing surface area[b]
SLST	268.7	kN	x	Thrust per engine (maximum continuous). CF6-80E1A2[h]
SPD	86 400	$sday^{-1}$		Time (Julian date) $\times SPD = $ Time (s)
T_0	288.15	K		Reference temperature (sea level)
Y_{pre}	2015	year	x	Present year for COC calculation
Y_{ref}	1993	year	x	Reference year for COC calculation
γ	1.4	−		Adiabatic gas constant

a Michael A. (2015); b Eurocontrol (2011); c EASA (2011); d ICAO (2005); e Penner et al. (1999); f IATA (2017); g MEW was estimated, because the exact value was unavailable; h EASA (2013); i Aircraft Commerce (2008); j Anthony (2009); k IMF (2016)

Table 2. Properties assigned to a resulting flight trajectory. The properties of the three groups (divided by rows) are obtained from the nearest grid point of EMAC at departure time of the flight, the flight trajectory calculation (Fig. 1), and the fuel-emissions-cost-climate calculation (Fig. 1; some properties are calculated in flight trajectory optimizations depending on a selected routing option), respectively. The attribute type indicates where the values of properties are allocated. "W", "S" and "T" stand for waypoints ($i = 1, 2, \cdots, n_{\mathrm{wp}}$), flight segments ($i = 1, 2, \cdots, n_{\mathrm{wp}} - 1$), and a whole flight trajectory in column 3, respectively. The column "New in V2.0" denotes properties newly introduced in AirTraf 2.0.

Property	Unit	Attribute type	New in V2.0	Description
$\mathrm{aCCF_{O_3}}$	$\mathrm{K(kg(NO_2))^{-1}}$	W	x	Algorithmic Climate Change Function of ozone[a,b]. See Eq. (A1)
$\mathrm{aCCF_{CH_4}}$	$\mathrm{K(kg(NO_2))^{-1}}$	W	x	Algorithmic Climate Change Function of methane[a,b]. See Eq. (A2)
$\mathrm{aCCF_{H_2O}}$	$\mathrm{K(kg(fuel))^{-1}}$	W	x	Algorithmic Climate Change Function of water vapor[a,b]. See Eq. (A3)
$\mathrm{aCCF_{CO_2}}$	$\mathrm{K(kg(fuel))^{-1}}$	W	x	Algorithmic Climate Change Function of CO_2[c]. See Eq. (A4)
$\mathrm{aCCF_{contrail}}$	$\mathrm{K(km(contrail))^{-1}}$	W	x	Algorithmic Climate Change Function of contrails[d]. See Eq. (A5)
Potcov	fraction	W	x	Potential persistent contrail cirrus coverage[e]
P	Pa	W		Pressure
T	K	W		Temperature
ρ	$\mathrm{kg m^{-3}}$	W		Air density
u, v, w	$\mathrm{m s^{-1}}$	W		Three dimensional wind components
a	$\mathrm{m s^{-1}}$	W		Speed of sound
$\mathrm{ATR20_{O_3}}$	K	S	x	Anticipated climate impact of ozone. $\mathrm{ATR20_{O_3}}_{,i} = \mathrm{aCCF_{O_3}}_{,i} \times \mathrm{NO}_{x,i} \times 10^{-3}$
$\mathrm{ATR20_{CH_4}}$	K	S	x	Anticipated climate impact of methane. $\mathrm{ATR20_{CH_4}}_{,i} = \mathrm{aCCF_{CH_4}}_{,i} \times \mathrm{NO}_{x,i} \times 10^{-3}$
$\mathrm{ATR20_{H_2O}}$	K	S	x	Anticipated climate impact of water vapor. $\mathrm{ATR20_{H_2O}}_{,i} = \mathrm{aCCF_{H_2O}}_{,i} \times \mathrm{FUEL}_i$
$\mathrm{ATR20_{CO_2}}$	K	S	x	Anticipated climate impact of CO_2. $\mathrm{ATR20_{CO_2}}_{,i} = \mathrm{aCCF_{CO_2}} \times \mathrm{FUEL}_i$
$\mathrm{ATR20_{contrail}}$	K	S	x	Anticipated climate impact of contrails. $\mathrm{ATR20_{contrail}}_{,i} = \mathrm{aCCF_{contrail}}_{,i} \times \mathrm{PCC}_{\mathrm{dist},i}$
$\mathrm{ATR20_{total}}$	K	S	x	Anticipated climate impact (total). See Eq. (8)
d	m	S		Flight distance
ETO	Julian date	W		Estimated time over
FT	s	T		Flight time. $\mathrm{FT} = (\mathrm{ETO_{n_{wp}}} - \mathrm{ETO_1}) \times \mathrm{SPD}$
h	m	W		Altitude
$\overline{h}$	m	T		Mean flight altitude. $\overline{h} = 1/n_{\mathrm{wp}} \sum_{i=1}^{n_{\mathrm{wp}}} h_i$ with waypoint number n_{wp}.
$\mathrm{PCC_{dist}}$	km(contrail)	S	x	Contrail distance[f]
V_{TAS}	$\mathrm{m s^{-1}}$	W		True air speed
V_{ground}	$\mathrm{m s^{-1}}$	W		Ground speed
λ	deg	W		Longitude
ϕ	deg	W		Latitude
COC	USDollar	T	x	Cash operating cost[g]
$\mathrm{EINO}_{x,a}$	$\mathrm{g(NO_x)(kg(fuel))^{-1}}$	W		NO_x emission index
F_{cr}	$\mathrm{kg(fuel) s^{-1}}$	W		Fuel flow of an aircraft (cruise)
FUEL	kg	S		Fuel use
$\mathrm{H_2O}$	$\mathrm{g(H_2O)}$	S		H_2O emission
m	kg	W		Aircraft weight
NO_x	$\mathrm{g(NO_x)}$	S		NO_x emission
SOC	USDollar	T	x	Simple operating cost

[a] Van Manen (2017); [b] Van Manen and Grewe (2019); [c] Dahlmann (2018); [d] Yin et al. (manuscript in preparation, 2019); [e] Frömming et al. (2014); [f] Yin et al. (2018a); [g] Liebeck et al. (1995)

Table 3. Setup for AirTraf one-day simulations. The setup of the two groups (divided by rows) are used for AirTraf/EMAC and for ARMOGA (Sasaki et al., 2002; Sasaki and Obayashi, 2004, 2005), respectively. α is an user-specified crossover parameter; r_m is a mutation rate; and η_m is an parameter controlling the shape of a probability distribution. Details of these parameters are described in Yamashita et al. (2016).

Parameter	Description
ECHAM5 resolution	T42L31ECMWF (2.8° by 2.8°)
Simulation period	December 1, 2015 00:00:00 − December 2, 2015 00:00:00 UTC
Time step of EMAC	12 min
Flight plan	103 trans-Atlantic flights (eastbound 52/westbound 51)[a]
Aircraft type	A330-301
Engine type	CF6-80E1A2, 2GE051 (with 1862M39 combustor)
Flight altitude changes	[FL290, FL410] (fixed at FL350 for the great circle option)
Mach number	0.82
Number of waypoints, n_{wp}	101
Design variable, n_{dv}	11 (6 locations and 5 altitudes)
Population size, n_p	100
Number of generations, n_g	100
Selection	Stochastic universal sampling
Crossover	Blend crossover BLX-0.2 ($\alpha = 0.2$)
Mutation	Revised polynomial mutation ($r_m = 0.1$; $\eta_m = 5.0$)

a Grewe et al. (2014a) and REACT4C (2014)

Table 4. The nine performance measures obtained from the one-day AirTraf simulations with different aircraft routing options (the values indicate the sum of 103 flights). The minimum values of each performance measure are marked with an asterisk; changes (in %) relative to the COC option are given in parentheses. Bar charts of the same data are given in Fig. S4 in the Supplementary material.

Routing option	Flight distance 10^3km	Flight time h	Fuel use ton	NO_x emission ton	H_2O emission ton	Contrail distance 10^3km	SOC Mil.USD	COC Mil.USD	ATR20$_{total}$ 10^{-7}K
Great circle	660.3* (−0.4)	757.4 (+0.1)	3979.1 (+5.8)	44.6 (+5.5)	4894.2 (+5.8)	154.9 (+19.1)	4.072 (+2.9)	5.463 (+2.1)	6.85 (+12.5)
Flight time	663.2 (+0.02)	739.4* (−2.3)	4521.9 (+20.2)	57.8 (+36.8)	5562.0 (+20.2)	127.7 (−1.9)	4.299 (+8.6)	5.673 (+6.1)	10.44 (+71.5)
Fuel use	663.3 (+0.03)	757.3 (+0.1)	3758.5* (−0.1)	42.2 (−0.2)	4623.0 (−0.1)	128.5 (−1.2)	3.960 (+0.03)	5.351 (+0.03)	5.85 (−3.9)
NO_x emission	664.5 (+0.2)	758.8 (+0.3)	3766.8 (+0.1)	42.1* (−0.5)	4633.1 (+0.1)	131.8 (+1.3)	3.968 (+0.2)	5.360 (+0.2)	5.83 (−4.2)
H_2O emission	663.3 (+0.03)	757.3 (+0.1)	3758.5 (−0.1)	42.2 (−0.2)	4623.0* (−0.1)	128.5 (−1.2)	3.960 (+0.03)	5.351 (+0.03)	5.85 (−3.9)
Contrail formation	717.4 (+8.2)	812.3 (+7.4)	4625.5 (+23.0)	57.0 (+34.9)	5689.3 (+23.0)	26.3* (−79.8)	4.549 (+14.9)	5.990 (+12.0)	3.45 (−43.4)
SOC	663.2 (+0.02)	756.6 (+0.03)	3760.4 (−0.02)	42.2 (−0.1)	4625.3 (−0.02)	130.2 (+0.1)	3.959* (0.0)	5.349 (0.0)	6.02 (−1.1)
COC	663.1	756.4	3761.1	42.3	4626.2	130.1	3.959	5.349*	6.09
Climate impact	703.2 (+6.0)	801.4 (+5.9)	4474.0 (+19.0)	52.3 (+23.8)	5503.1 (+19.0)	92.6 (−28.8)	4.443 (+12.2)	5.874 (+9.8)	1.96* (−67.9)