

# Performance and applicability of a 2.5D ice-flow model in the vicinity of a dome

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**Abstract.** Three-dimensional ice flow modelling requires a large amount of computing resources and observation data, such that 2D simulations are often preferable. However, when there is significant lateral divergence, this must be accounted for (2.5D models), and a flow tube is considered (volume between two horizontal flow lines). In the absence of velocity observations, this flow tube can be derived assuming that the flow lines follow the steepest slope of the surface, under a few flow assumptions. This method typically consists of scanning a digital elevation model (DEM) with a moving window, and computing the curvature at the center of this window. The ability of the 2.5D models to account properly for a 3D state of strain and stress has not clearly been established, nor their sensitivity to the size of the scanning window, and to the geometry of the ice surface, for example in the cases of sharp ridges. Here, we study the applicability of a 2.5D ice flow model around a dome, typical of the East Antarctic plateau conditions. A twin experiment is carried out, comparing 3D and 2.5D computed velocities, on three dome geometries, for several scanning windows and thermal conditions. The chosen scanning window used to evaluate the ice surface curvature should be comparable to the typical radius of this curvature. For isothermal ice, the error made by the 2.5D model is in the range 0-10% for weakly diverging flows, but is 2 or 3 times higher for highly diverging flows and could lead to a non-physical ice surface at the dome. For non-isothermal ice, assuming a linear temperature profile, the presence of a sharp ridge makes the 2.5D velocity field unrealistic. In such cases, the basal ice is warmer and more easily laterally strained than the upper one, the walls of the flow tube are not vertical, and the assumptions of the 2.5D model are no longer valid.

## 1 Introduction

Computing performance has continued to increase through the last decade, and 3D numerical simulations that could not be performed a few years ago are nowadays affordable. In particular, the Stokes equations can be solved on large 3D datasets (e.g., Gillet-Chaulet et al., 2012), such that the complete state of stress is accounted for. Nevertheless, two-dimensional ( $x, z$ ) models are still very useful, since they are easier to handle, the computing time is at least two orders of magnitude less, and they need fewer observations to describe the boundary conditions. Two-dimensional models assume to be unchanging in the transverse direction, such that they apply well when the topography is always the same in the missing  $y$ -direction. This is a reasonable assumption for the case of large ice streams, or on the flanks of a symmetric ice-cap (e.g., Martín et al., 2006).

Under this assumption, the ice undergoes a strain in a vertical plane only, and all the components of strain in the transverse direction vanish (*plane strain* case).

In this study, we describe the volume between theoretical stream lines as a "flow tube". (Fig. 1, red lines). If the ice flow is locally converging or diverging, 2D models are no longer valid, as they do not conserve the mass in this case, and cannot account for lateral mechanical stresses imposed by the width variations of the flow tube. For example, Sergienko (2012) showed that the 2D ( $x, z$ ) models could not account for the flow of ice on a bumpy bed, because they consider no lateral variability, and recommended to use such models on a width-averaged topography. It is however possible to account for variations in the width of the flow tube in the model, and still express the equations with ( $x, z$ ) coordinates only (Reeh, 1988); we hereafter call these models 2.5D models. Under a few flow assumptions (the bed elevation varies slowly in the transverse direction, the walls of the flow tube are vertical, the lateral curvature of the flow line is small), such models represent an improvement on 2D models because they account for lateral variability, while maintaining the computational speed which full 3D models lack.

A particularly simple case is that of an axisymmetric circular dome, where the width of the flow tube increases linearly along the flowline. In any other case, there are several possible methods for determining the width of the flow tube. At a large scale, the ice velocity can be determined via interferometric synthetic-aperture radar data (Rignot et al., 2011), but this is only reliable when ice velocity is large compared to the associated error. In the interior of ice sheets, where the ice velocity is low, velocity measurements can be made using stake monitoring. If correctly positioned, the direction and position of the stakes may give the width of the flow tube (Waddington et al., 2007). Without any available velocity observations, the only possibility is to use a digital elevation model (DEM) describing the shape of the surface. The velocity measurement method is preferable when possible, since the ice does not always flow perpendicular to the surface contour lines. However, along a divide, or along the centre line of a drainage basin, the longitudinal surface velocity reaches its minimum or its maximum value compared to the lateral surroundings; moreover, if the flow line has a slight horizontal curvature, the horizontal shear strain rate can be neglected (Reeh, 1988). The ice velocity is then oriented along the steepest surface slope and the curvature of the surface contour lines linked to the width of the flow tube; for a dome or a two-dimensional ridge, determining the curvature of the surface contour line is sufficient to describe the widening of the flow tube. In this case, to determine the local curvature of a surface, the neighborhood of each point of a DEM is approximated via interpolation. The size of this neighborhood (scanning window) can be freely chosen, but no objective criteria are currently available to make this choice. This approach has been used to compute the surface curvature of an ice sheet (Rémy and Minster, 1997; Rémy et al., 1999) but the size of the scanning window was not discussed. Due to the noise affecting the DEM and to the regional curvature, the local computed curvature may be ambiguous.

The 2D models that account for the width of the flow tube vary in the complexity of their physics and their underlying assumptions (e.g. Martín et al., 2009; Koutnik and Waddington, 2012; Sergienko, 2012). The 2.5D model proposed by Reeh (1988) assumes that the streamlines are perpendicular to the surface contour lines (Fig. 1), so that the mass conservation equation directly depends on the radius of curvature of the surface contour lines (here called  $R$ ). The model imposes a vertical profile for the horizontal velocity, by the use of a shape function, and the momentum conservation equation was thus not given for any case. This vertical profile is assumed to be unchanging through the  $x$ -direction (column-flow model). Later on,

Hvidberg (1996) improved upon this approach by setting the stress equilibrium equations without any assumption on the shape of the velocity field. Additionally, when considering a width-varying flow tube, the heat equation should be modified as well, but only few authors have considered this as necessary for their purpose (Hvidberg, 1993, 1996; Pattyn, 2002).

Different authors have used the modelling approach proposed by Reeh (1988), or similar ones, at different scales and for various geometries : for example, mountain glacier (Salamatin et al., 2000; Pattyn, 2002), ice-sheet domes (Reeh and Paterson, 1988; Hvidberg et al., 1997a), ice-sheet ridges (Hvidberg et al., 2002) or even for the whole Greenland (Reeh et al., 2002). The shape and size of the ice bodies are quite diverse, but unfortunately the validity of the 2.5D approach, and its assumptions, for these different cases has received little attention. In particular, the transverse deformation of the ice is assumed to be constant through depth (i.e. the walls of the tube are vertical), which may depend on the surface geometry. Furthermore the error in the computation of  $R$  is only discussed by Hvidberg et al. (1997b), who estimated the error in the calculation of  $R$  to be 15 %, and Hvidberg et al. (2001) by about 50 %. The method used to measure  $R$  is not detailed either, and we doubt that its influence is negligible.

No assumptions of the 2.5D model specifically prohibit its use for a highly diverging tube. However, (Reeh, 1989) determined that the model was incapable of handling such a flow regime, but this was established for a model assuming a constant horizontal velocity profile. Furthermore, Reeh (1989) accounted for the spatial evolution of  $R$  with a simple linear model, based on the surface observation. These considerations suggest that domain of applicability of 2.5D models with regards to flow divergence and surface geometry remains to be determined.

As a consequence, the applicability of the 2.5D model should particularly be examined on dome geometries, where simple 2D models would be unable to account for flow divergence. The flow tubes may widen by several orders of magnitude on a few tens of kilometers, especially on the sharpest ridge of the dome. The goal of this study is to perform several comparisons between 2.5D and 3D models, for various dome geometries and temperature conditions, to answer the following questions:

1. What is the error associated with the computation of  $R$  from the DEM? This geometric error depends largely on the size of the scanning window.
2. How well can a 3D state of stress be accounted for by the single parameter  $R(x)$  along the flow line? This is related to the inherent error of the 2.5 model and its underlying assumptions.

To investigate the performance of the 2.5D model, we compare the velocity fields resulting from the 2.5D and 3D models, the latter being taken as a reference. To our knowledge, no such a systematic comparison between the results of 3D and 2.5D models has been carried out. As such, we hope the present study will guide future research using 2.5D models in different scenarios. In the following, we present the equations of the 3D- and the 2.5D model. We then run the simulations on several domes of different shapes: a circular one (axisymmetric), a slightly elongated one, and a very elongated one. We first consider the isothermal case, before moving on to investigate the effects of temperature.

## 2 Description of the 3D model

In future work, we hope to investigate a small dome on the East Antarctic plateau. As such, we presently consider a synthetic case with similar geometric and thermal conditions.

### 2.1 Geometry and mesh

5 In order to investigate the influence of flow divergence, we model a ridge of a dome, as this results in significant divergence. We perform the present model comparison on a synthetic geometry which consists of a 15 km-radius domain, whose shape is a quarter of cylinder only, for reasons of symmetry. The initial thickness of the ice is 3239 m at the summit, the mean surface slope is around 0.6/1000 and the underlying bed is flat. The space coordinate is a  $(x, y, z)$  cartesian system.

The 3D mesh is horizontally unstructured and vertically extruded on 10 levels. The horizontal mean spacing between the  
10 nodes is 1 km (Fig. 2).

### 2.2 Mechanical model

#### 2.2.1 Conservation equations

We denote the velocity vector  $\mathbf{u}$ , with components  $(u, v, w)^t$ . The stress and strain rate tensors are denoted  $\boldsymbol{\sigma}$  and  $\dot{\boldsymbol{\epsilon}}$  respectively, and their components,  $\sigma_{ij}$  and  $\dot{\epsilon}_{ij}$ . The deviatoric part of  $\boldsymbol{\sigma}$  is denoted  $\boldsymbol{\tau}$ , and its components,  $\tau_{ij}$ . The 3D mechanical model  
15 consists of a Stokes problem for incompressible ice of density  $\rho$ , in which the mass and momentum conservations equations are written

$$\nabla \cdot \mathbf{u} = 0 \tag{1}$$

$$\nabla \cdot \boldsymbol{\sigma} + \rho \mathbf{g} = 0 \tag{2}$$

where  $\mathbf{g}$  is the gravitational acceleration vector. The values of the different parameters are given in Table 1. The ice is assumed  
20 to deform following Glen's generalized flow law (Glen, 1958):

$$\dot{\epsilon}_{ij} = A(T) \tau_e^{n-1} \tau_{ij} \tag{3}$$

where  $\tau_e$  is the second invariant of  $\boldsymbol{\tau}$ . We choose a value of  $n = 3$ . The rate factor  $A(T)$  non linearly depends on temperature, following an Arrhenius law (Cuffey and Paterson, 2010), reflecting the fact that warm ice is softer. In Antarctica, the temperature range between the surface and the bottom can be large, and the pressure melting point is frequently reached at the bedrock.  
25 As a consequence, the viscosity of the upper ice can be two orders of magnitude larger than near the bedrock. To study the influence of the temperature  $T$  (expressed in kelvin) on the performance of our 2.5D model, we first consider isothermal ice at 245 K, and then a non-isothermal ice, for which  $T(z) = 270 - 50 \cdot (z - b)/(s - b)$ , where  $s$  and  $b$  are the elevation of the surface and the bedrock, respectively. This linear temperature profile is simple but realistic enough to show the effect of a warmer ice at the bottom. For convenience, and as the bed is flat,  $b = 0$  in the following experiments.

## 2.2.2 Boundary conditions

Since the 3D mesh is a quarter of a cylinder, the conditions have to be set on 5 different boundaries, numbered from BC1 to BC5 (Fig. 2). We consider a frozen ice at the bed and no flow through the lateral boundaries (symmetry condition). Considering no sliding at the bottom and neglecting the atmospheric pressure at the surface, the boundary conditions are written as follows:

$$5 \quad BC1 : \mathbf{u} \cdot \mathbf{n}|_{y=0} = 0 \quad (4)$$

$$BC2 : \mathbf{u} \cdot \mathbf{n}|_{x=0} = 0 \quad (5)$$

$$BC3 : \mathbf{u}|_{z=0} = \mathbf{0} \quad (6)$$

$$BC4 : \boldsymbol{\sigma} \cdot \mathbf{n}|_{z=s} = 0 \quad (7)$$

where  $\mathbf{n}$  is the outward facing normal vector on the surface. Since the surface (BC4) is let free, a kinematic boundary condition  
 10 for the surface  $s(x, y, t)$  has to be solved as well:

$$\frac{\partial s}{\partial t} + u \frac{\partial s}{\partial x} + v \frac{\partial s}{\partial y} = w + a \quad (8)$$

where  $a$  is the accumulation rate. The velocity boundary condition on BC5 is set to control the shape of the steady-state dome. The method to create an elongated dome is similar to that of Gillet-Chaulet and Hindmarsh (2011), in which the shallow ice approximation is used to prescribe a profile of horizontal velocity at the boundary. As such, BC5 is not a physical boundary, but  
 15 it allows the domain to be reduced to a reasonable extent. The velocity vector is oriented along the surface slope, and pointing outwards. It is variable with depth to the power  $n + 1$  and vanishes at the bed. Its norm is tuned depending on its orientation, so that the shape of the dome can be elongated in one preferred direction (Fig. 3):

$$u = \bar{w} \frac{2 \cos \theta}{\alpha} \frac{(n+2)}{(n+1)} \times \left( 1 - \left( 1 - \frac{z}{s} \right)^{(n+1)} \right) \quad (9)$$

$$20 \quad v = \bar{w} \frac{2(\alpha-1) \sin \theta}{\alpha} \frac{(n+2)}{(n+1)} \times \left( 1 - \left( 1 - \frac{z}{s} \right)^{(n+1)} \right) \quad (10)$$

where  $\theta$  is the angle to the edge of the domain, and  $\alpha$  a shape parameter controlling the elongation of the dome. The following results will correspond to a stabilized steady-state geometry, for a uniform and constant accumulation  $a$  in time and space. To do so, the output mean velocity  $\bar{w}$  is tuned to balance the surface accumulation, so that it can be expressed as

$$\bar{w} = \frac{a \cdot \Sigma}{W_L \cdot s} \quad (11)$$

25 where  $\Sigma$  is the surface area of the accumulation zone on the top boundary (gray area in Fig. 1), and  $W_L$  the width of the ice flow at the downstream position  $x = L$ . In this particular case,  $\Sigma$  and  $W_L$  are simply equal to 1/4 of the dome surface and dome

perimeter, and  $L$  is the radius of the dome. Three different values were taken for the shape parameter  $\alpha$ : 2 (axisymmetric case, circular geometry), 3, and 6 (Fig. 3). The axisymmetric domain will test the performance of the model for a perfectly known case. The latter two cases will test the ability of the model to account for divergence of varying magnitudes.

### 3 Description of the 2.5D model

5 The coordinate system used by Reeh (1988) is a curvilinear coordinate system with right-handed oriented coordinate axis. The  $x$ -axis is oriented along the flow line, the  $z$ -axis is vertically oriented, and the  $y$ -axis is transverse to flow, and tangential to a surface contour line. As we only consider here straight flow lines (linear ridge of an ice divide), the coordinate system is locally cartesian (Fig. 1). We refer to the 2.5D model in the  $(x, z)$  coordinate system. We now recall the assumptions made by Hvidberg (1996), partly inherited from Reeh (1988).

- 10
1. The flowlines are perpendicular to the surface contour lines.
  2. The direction of the horizontal velocity components are constant with depth, which implies that the walls of the flow tube are vertical.
  3. There is no shear stresses on the vertical boundaries defined by the flow tube.
  4. The ice deforms according to Glen's flow law.

15 These assumptions together mean that the surface horizontal strain is transferred to the bottom, so that the surface contour lines and the horizontal velocity in the flow direction impose the transverse stresses. Such assumptions are reasonable in the center of an ice-sheet for a slowly varying bed (Reeh, 1988). If the bedrock spatial variations are too steep, they will warp the ice free surface so that the velocity at the base may not be parallel to the velocity at the surface (Hvidberg, 1993; Sergienko, 2012).

#### 20 3.1 Geometry

The 2D domain is taken as a vertical slice of the 3D domain, on one of its lateral boundaries (Fig. 2). The dome being elongated, we will run the 2.5D model along the sharpest ridge of the 3D dome ( $y = 0$ ) or perpendicular to it ( $x = 0$ ).

#### 3.2 Mechanical model

25 We denote the width of the considered flow tube  $W(x)$ . The radius of the surface contours lines  $R(x)$  is taken positive for diverging flow and negative for converging flow. In this model, the assumption (2) implies that  $W$  has no dependence on  $z$ , so that geometrical considerations show that  $W(x)$  is directly linked to  $R(x)$  by

$$\frac{1}{R(x)} = \frac{1}{W(x)} \frac{\partial W}{\partial x} \tag{12}$$

An axisymmetric dome leads to the simple relations  $R(x) = x$  and  $W(x) \propto x$ . If the flow tube is diverging more than the axisymmetric flow (on a ridge for example), the corresponding tube surface is narrowed *for a given output width*, and leads to lower output velocities.

The following sections present the equations of mass and momentum conservation, modified to account for the divergence of the tube. As the velocity field mainly depends on the input/output balance, some authors only conserve the mass (e.g., Parrenin et al., 2004; Todd and Christoffersen, 2014), but not the momentum; this approach can in particular be sufficient for a vertically-integrated model (Hvidberg et al., 1997b). Other authors do not modify the Stokes equations at all, but instead add an extra-surface mass balance term (Cook et al., 2014; Gladstone et al., 2012) which depends on the divergence of the tube. This approach has the advantage of simplicity and results in a correct output flux, but neglects the true horizontal advection of the ice. However, this can be justified for ice sheet margins, where the ice mainly undergoes sliding. For all these mass-only conservation models, the normal lateral stress of the surrounding ice is not accounted for, since the force equilibrium is not properly modified.

### 3.2.1 Mass conservation

At the ice divide, the velocity component  $v$  and its spatial derivatives vanish for reasons of symmetry, so that there is no dependence of the strain rates on the transverse coordinate. Under the above assumptions and considering a flow tube of width  $W(x)$  (and corresponding radius  $R(x)$ ), the normal strain rates in the curvilinear system are then written (Jaeger, 1969, p. 45):

$$\dot{\epsilon}_{xx} = \frac{\partial u}{\partial x}; \dot{\epsilon}_{yy} = \frac{u}{R(x)}; \dot{\epsilon}_{zz} = \frac{\partial w}{\partial z} \quad (13)$$

If the flow tube has a constant width, the value of  $R$  is infinite and the equation correspond to the plane strain case. For the more complete form of these expressions, see the discussion of Reeh (1988). The mass conservation then follows:

$$\frac{\partial u}{\partial x} + \frac{u}{R(x)} + \frac{\partial w}{\partial z} = 0 \quad (14)$$

### 3.2.2 Momentum conservation

For a straight flowline, the horizontal shear strain rate is written (Reeh, 1988):

$$\dot{\epsilon}_{xy} = \frac{1}{2} \left( \frac{\partial u}{\partial y} - \frac{v}{R(x)} + \frac{\partial v}{\partial x} \right) \quad (15)$$

Along a divide between drainage basins,  $v$  vanishes and  $u$  attains a local maximum so that  $\partial u / \partial y = 0$ . In this particular case, there is no horizontal shear strain, and the only non-zero components of strain are the normal ones and the vertical strain rate  $\dot{\epsilon}_{xz}$ . For dome geometries, Hvidberg (1993) shows that the curvilinear coordinate system is equivalent to a cylindrical coordinate system distorted in the  $y$ -direction, for which the radius of curvature  $R(x)$  describes the local distortion. The force equilibrium equations, expressed in the  $(x, z)$  cartesian coordinate system, are inherited from their formulation in cylindrical coordinates, and are written (Hvidberg, 1996; Jaeger, 1969, p. 123):

$$\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{xz}}{\partial z} + \frac{\sigma_{xx} - \sigma_{yy}}{R(x)} = 0 \quad (16)$$

$$\frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{zz}}{\partial z} + \frac{\sigma_{xz}}{R(x)} = \rho g \quad (17)$$

where  $\sigma_{yy}$  is known in terms of  $u$  and  $R(x)$  (Eqs. (3) and (13)).

### 3.2.3 Boundary conditions

- 5 The boundary conditions are inherited from the 3D case: no sliding, free surface, vanishing velocity at the ice divide and an imposed horizontal velocity profile downstream. Note that the value of the mean output velocity  $\bar{w}$  directly depends on  $\Sigma$ , which is now given by

$$\Sigma = \int_0^L W(x) dx \quad (18)$$

- 10 where  $L$  is now the length of the flow line. Equations (11) and (18) together mean that the errors in the calculation of  $W$  result in errors in the prescribed output velocity. Since we consider a straight flow line, there is no transverse flow across the considered plane. The free surface equation is thus derived from Eq. (8), and is equivalent to a simple 2D case, here given for the ( $y = 0$ ) plane:

$$\frac{\partial s}{\partial t} + u \frac{\partial s}{\partial x} = w + a \quad (19)$$

### 3.3 Implementation in Elmer/Ice

- 15 The modified mechanical equations are implemented in the Elmer/Ice finite element software (Gagliardini et al., 2013). The correct implementation of the mass conservation was checked by comparing different 2.5D simulations with the Vialov-type profiles (Vialov, 1958) computed for different diverging tubes. The expression of the Vialov profile in the case of a power-law varying flow tube is presented in Appendix A.

### 3.4 Determination of the contour radius

- 20 To determine the radius of curvature of the surface contour lines, we first export a DEM from the surface nodes of the 3D model. These nodes are fitted using an inverse distance weighting, with a power of 4 in order to ensure a good smoothing of the computed surface, representative of a real ice sheet. For comparability with a real case, the spatial resolution of the DEM is taken equal to 400 m, which is the resolution of the DEM resulting from the ICESat mission on Antarctica (Schutz et al., 2005). The DEM surface curvature is computed for each pixel, using a scanning window for fitting (Fig. 1). The local surface  
25 inside the window is curve-fitted by a bivariate quadratic polynomial function, and the analytical curvature of this function taken as the local curvature, which depends on the size of the scanning window used. For example, along an elongated ridge



(Fig. 3, middle and right), the surface contours are close to ellipses, but only a sufficiently large scanning window will be able to account for this global shape. A small scanning window may compute an approximately circular curvature. On the contrary, even though a large window may lead to a more accurate value of the curvature geometrically speaking, it is not necessarily the case that it will also lead to a more accurate velocity field, since it is difficult to know which surrounding environment is mainly influencing the ice flow. As a consequence, three different sizes of scanning window are here tested: 2.8 km, 6 km and 10 km, thus corresponding to a width of 7, 15 and 25 pixels. For each pixel, the value of  $R(x)$  is taken as the inverse of the curvature of the contours of the fitted surface in the center of the window. This is done using GRASS GIS software.

### 3.5 Protocol of comparison

We first run the 3D transient isothermal simulation, and stop when a steady state is reached ( $\partial s/\partial t < 10^{-6}$  m/a). We then use the resulting DEM to compute the profile of  $R$  to initialize the 2.5D model. For each dome configuration ( $\alpha = 2, 3, 6$ ) we compare the different runs chosen: a) 3D, b) 2.5D for the three scanning windows, and a fixed geometry, c) 2.5D for the three scanning windows, with a free surface. For the axi-symmetric case we add a true 2.5D axisymmetric run (imposing an analytical value  $R = x$ ). Then we compare the 3D and 2.5D results for a non-isothermal ice, to investigate the influence of the temperature, especially near the base of the ice sheet.

## 4 Results and discussion

### 4.1 Circular geometry ( $\alpha = 2$ ), isothermal ice

#### 4.1.1 3D/2.5D axisymmetric comparison

The absolute error on the ice velocity for an axisymmetric 2.5D model ( $R = x$ ) is of the order of  $10^{-4}$  m/a (Fig. 4.a, where the black and yellow curves are almost superimposed). The observed error is a result of discretization, and should tend to zero as the element sizes decrease.

#### 4.1.2 3D/2.5D comparison

The computation of the radius of the surface contour lines is strongly influenced by the size of the scanning window. For a circular geometry, the variation of  $W$  along  $x$  should be linear, which is almost the case for the two wider windows (Fig. 5, solid lines). With this regular geometry, the larger the window, the more precise the radius, since the fitted surface will be more accurate. On the contrary, the width value computed with the smaller window is less regular and underestimated by about 30%, meaning that we cannot evaluate a certain curvature from too small a sample. When choosing the intermediate or the larger scanning-window, the root mean square error (RMSE) in the velocity is 9.9 % or 3.1 % respectively (Fig. 4.a), and is a consequence of the error in the calculation of the radius.

## 4.2 Elongated domes ( $\alpha = 3$ and $\alpha = 6$ ), isothermal ice

For elongated domes, we consider both the flow line along the sharpest ridge ( $y = 0$ ), and perpendicular to it ( $x = 0$ ).

### 4.2.1 3D/2.5D comparison, along the ridge

Along a ridge, the flow tube is non linearly diverging. For a given output width, the accumulation area is smaller than in the axisymmetric case, thus leading to lower output velocities.

With fixed geometries, it clearly appears that the velocity is underestimated for elongated domes (Fig. 4.b and 4.c, dashed lines), meaning that the local surface slope cannot explain the ice motion by itself: the ice along the ridge is also, if not mainly, pulled by the surrounding lateral ice, which moves due to a steeper surface slope. The case of the small scanning window appears to be different (Fig. 4.b, red dashed line), simply because of a really poor estimation of  $W$ , thus of the output velocity.

The downstream velocities are always quite accurate (10 % error), since they mainly depend on the tube surface calculation, incorporated in the velocity boundary condition. When releasing the surface, the surface slope slightly increases to accommodate the velocity boundary condition, and the computed velocity field is then closer to the 3D reference. The relative error made in the downstream part of the flow is comparatively higher near the divide since the velocities are very small.

In case of a sharp ridge, the ice surface has the shape of a circus tent (Fig. 6, dotted line), i.e. the ice slope increases when going towards the divide. This shape is clearly not physical, and is the result of a numerical artefact. Since the vertical strain rate is always of the order of  $a/H$ , the conservation equation leads to a balance between  $\partial u/\partial x$  and  $u/R$  close to the divide. For highly diverging flows,  $u/R$  is much smaller than for an axisymmetric case at the same  $x$  position (Appendix B). As a consequence,  $\partial u/\partial x$  should have at a comparatively much higher value. The only way for  $u$  to increase over a short distance near the divide is for the surface elevation to decrease sharply. To handle this artefact, we increased the mesh resolution near the dome, but without any successful results. This artefact does not appear for  $\alpha = 3$  (slightly elongated dome).

For  $\alpha = 6$ , the tube surface is better estimated with an intermediate window, whose size is closer to the local value of  $R$ . The RMSE in the velocity is 12.1 % for the intermediate window and 44.3 % for the large window. Too large a window would consider the whole shape of the dome and lead to an underestimation of  $R$ . The amplitude of the error between the different runs show that for sharp ridges (or highly diverging tubes) the choice of the window size is not straightforward, as a wider window increases the regularity of the velocity field, but decreases the ability to capture the local curvature.

Reeh (1989), using a 2.5D model for dating purposes, explained the large errors for the diverging tube of Camp Century partly by the simplicity of his model, especially his linear model of  $R$ . The computed velocities show important discrepancy with the observed ones, which leads to a bad estimation of the origin position of the ice, by 200 %. The error made in the velocity field with this more complete model seems now to be small enough for tracking ice particles correctly, for example with the intermediate window size. However, for a real case of highly curved surface, it is difficult to know *a priori* the best window size to use and if the divergence can be properly accounted for, in the absence of a reference as it is done here.

### 4.2.2 3D/2.5D comparison, perpendicular to the ridge

As the divergence is much smaller perpendicular to the sharpest ridge, the velocity field is much smoother, and its spatial evolution closer to the 3D reference than the along-ridge case. The RMSE is 11.9 % and 7.5 % for the intermediate and large window size respectively, for  $\alpha = 3$ , and 13.7 % and 11.1 % for  $\alpha = 6$ ; this error is slightly higher than for the circular geometry (Fig. 4.d). In the case of large radius values, towards the exterior of the dome, the wider window gives the more accurate results.

### 4.3 Non-isothermal ice

A supplementary comparison is carried out on the sharp ridge ( $y = 0$ ) for temperature varying linearly through depth. The computed velocity field towards the divide (low  $x$  values) shows a reversed vertical profile, i.e. the basal ice goes faster than the upper ice (Fig. 7, bottom). This non-physical result in 2.5D can be explained this way: as soon as a 3D tube diverges more than for an axisymmetric flow, the warmer basal ice is more easily laterally strained than the colder surface ice. As a consequence, the walls of the 3D flow tube are no longer vertical, and using the 2.5D model in such a case would violate assumption (2). This effect is particularly pronounced close to the divide, where the tube is narrower, and can be seen in the 3D simulations as follows (Fig. 8). The stream lines going through a flux gate at  $x = 1000$  m are tracked on a few hundred metres ( $x = 1200$  m for  $\alpha = 6$ ,  $x = 1500$  m for  $\alpha = 3$ ). The divergence of the stream lines is depending on their depth – the tube is larger near the bottom, Fig. 8, blue curves –, and this dependence is stronger for high diverging tubes. To accommodate the lateral strain in this case, the 2.5D model computes high horizontal velocities in the bottom, whereas the real motion is in fact mainly laterally oriented. No mesh refinement has been able to correct this problem. Since it does not happen for a constant temperature, it certainly originates from the lower viscosity of the basal ice (Fig. 8, red curves).

This result also suggests that, on sharp ridges and with non-isothermal ice, working with a fixed vertical profile of velocity will prevent from such unintended behaviour. This artefact may affect the results of Hvidberg et al. (2002), who study a flow line between GRIP (Summit) and North GRIP, because their model accounts for temperature. Their flowline stretches along a ridge which can be quite sharp, and for which the flow divergence is probably higher than axisymmetry. However, it is much longer than ours, and the ridge is the sharpest far from the summit, whereas it is the contrary in our case. The artefact that appears in our simulations may not be as sensitive in their case. Nevertheless, care must be taken in such cases, since the basic assumptions may not be justifiable and the model is likely to be outwith its application domain.

For reasons of continuity, the walls of the flow tube cannot be vertical in the direction perpendicular to the ridge either, but the effect is too weak to impact the computed velocity field.

## 5 Conclusions

A systematic comparison between 2.5D- and 3D models has been presented in order to evaluate the ability of the former to accurately compute the velocity field on a small dome of an ice sheet. The error made when estimating the value of the radius of the surface contour lines is of the order of 10% if the computation window is well chosen, though it can be comparatively

higher close to the divide. The radius of curvature of the surface elevation contour lines should be determined with a sufficiently large computation window, but choosing the optimum size is not completely straightforward; in any case we suggest it should not be less than one-third of the maximum measured radius, and several windows should be tested to ensure the robustness of the results.

5 The 2.5D model can be used without any specific restriction for tubes diverging less than and up to an axisymmetric flow. For isothermal ice, the model can be used with tubes diverging more than an axisymmetric flow, if the divergence is not too high. For very high divergence, the ice is in our study mainly pulled by the output boundary condition, and the resulted velocity field may be somewhat irregular, and surface geometry unphysical close to the divide. This means that, in the case of a sharp ridge, accounting for a certain state of stress via a single parameter  $R(x)$  is clearly not possible. In practice, the examples of  
10 surface topographies given in this study may serve as a reference to evaluate if using a 2.5D model for a real case is justifiable.

For non isothermal ice, the tube should not diverge more than axisymmetry, because the softer basal ice would be much more easily laterally strained in the case of an elongated dome. The walls of the flow tube are therefore not vertical, which violates the model assumptions, and the corresponding horizontal velocity profile may be not physical near the divide. This has significant consequences for dating purposes: as the computed velocity field shows too small values in the upper layers, the age  
15 of the ice is overestimated, and the modelled isochrone layers are too high. In the absence of reference 3D solution as produced here, the velocity field may not necessarily appear to be unphysical, but this does not mean that the numerical artefact does not significantly affect the age calculation.

This study shows that the use of a 2.5D model, which is a trade off between 2D and 3D, must avoid several pitfalls, and its applicability domain appears to be significantly narrower than initially thought. The geometric error, resulting from  
20 miscalculation of  $R$ , should have less impact on the quality of the results than the inherent errors related to the violation of the 2.5D model's assumptions, especially when the temperature varies through the ice column. The 2.5D model was designed for divides between drainage basins, but real flowlines along sharp divides should not be modelled this way. Finally, for a real case, it has not been established how flat the bedrock should be to ensure that the assumption of verticality of the walls of the flow tube is reasonably respected, even for a slightly divergent flow. The applicability domain of the 2.5D model for slightly  
25 horizontally curved flowlines has not been investigated here, but further uncertainties are expected in this case, for which some secondary-order terms are neglected in the mass conservation.

### **Code availability**

The presented simulations were performed using the finite element model Elmer/Ice v.7.0 rev. 7016. The source code of the 2.5D model is available in the distribution since v.8.0 rev. d9d4a2f, implemented in the AIFlow solver.

## Author contribution

The experiments were designed by OG, FP and JT. OP carried them out, helped by CR for analytical developments, and by FGC for the Elmer/Ice implementation. OP prepared the manuscript with contributions from all co-authors.

## Appendix A: Vialov profile for a power-law diverging tube

- 5 To check the correct implementation of the mass conservation in the 2.5D model, we hereafter compute the height of a Vialov profile corresponding to a regularly diverging flow tube. Note that such a surface is only representative of a single flow line, and not for a whole surface, as what is usually done for a Vialov profile in plane strain (Vialov, 1958) or axisymmetry (Ritz, 1992).

Figure 5 shows that we may approximate the shape of the flow tube by a power-law depending on the  $x$ -coordinate. Let  
 10 consider a flow tube of width  $W = W_L \left(\frac{x}{L}\right)^\beta$ . For plane flow,  $\beta = 0$ , for axisymmetry  $\beta = 1$ , and for sharp ridges  $\beta > 1$ . The volume outflow  $q^*$  for a certain coordinate  $x$  may be expressed in one of two ways (Cuffey and Paterson, 2010, p.388):

$$q^* = \int_0^x a \left(\frac{x'}{L}\right)^\beta W_L dx' = \frac{ax}{(\beta+1)} \left(\frac{x}{L}\right)^\beta W_L = \frac{axW}{(\beta+1)} \quad (\text{A1})$$

$$q^* = \left(\frac{2A}{n+2} \tau_b^n H\right) HW \quad (\text{A2})$$

- 15 where  $H$  is the ice thickness,  $\tau_b$  the basal shear,  $a$  the accumulation rate,  $L$  the length of the glacier,  $A$  the rate factor of Glen's flow law. Equating the two expressions yields

$$a \cdot x = \frac{2A(\beta+1)}{n+2} \tau_b^n H^2 \quad (\text{A3})$$

The following reasoning is then similar to that of Cuffey and Paterson (2010), simply modified by a  $(\beta+1)$  multiplier. The final expression for the ice thickness is unchanged

$$20 \quad H = H^* \left(1 - \left(\frac{x}{L}\right)^{\frac{n+1}{n}}\right)^{\frac{n}{2n+2}} \quad (\text{A4})$$

except the height of the ice sheet  $H^*$  at the dome ( $x = 0$ ), which is now

$$H^* = \left(\frac{2(n+2)^{1/n}}{\rho g}\right)^{\frac{n}{2n+2}} \sqrt{L} \left(\frac{a}{2A(\beta+1)}\right)^{\frac{1}{2n+2}} \quad (\text{A5})$$

This expression is consistent with the one previously derived for axisymmetry. We then use this expression to control the 2.5D model by comparing the value of  $H^*$  computed by the model with its above theoretical value.

## Appendix B: Radius and surface of a power-law diverging flow tube

We consider the same flow as in Appendix A. The value of the radius  $R(x)$  is then expressed as

$$\frac{1}{R(x)} = \frac{1}{W} \frac{dW}{dx} = \frac{d \ln(W)}{dx} = \frac{\beta}{x} \quad (\text{B1})$$

The surface area of the tube upstream of  $x$  can be expressed as

$$5 \quad \Sigma(x) = \int_0^x W(x) dx = \frac{W_L}{\beta + 1} \frac{x^{\beta+1}}{L^\beta} \quad (\text{B2})$$

As  $u$  is more or less proportionnal to the upstream surface area  $\Sigma$ ,  $u/R$  is expected to be proportionnal to  $W_L (\frac{x}{L})^\beta$ . On the contrary, one can consider that the value of  $\partial u / \partial x$  should be of the order of  $\bar{\omega}/L$ , i.e. simply proportionnal to  $\Sigma(L)/L = W_L/(\beta + 1)$ . Near the divide,  $u/R$  is then comparatively much smaller than  $\partial u / \partial x$  for sharp ridges than for axisymmetric flows, and imbalances the corresponding mass conservation.

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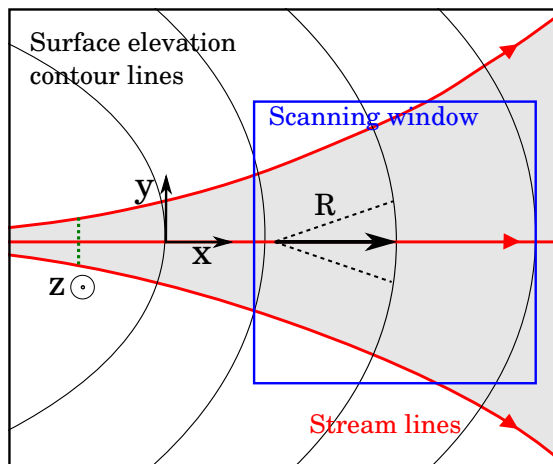
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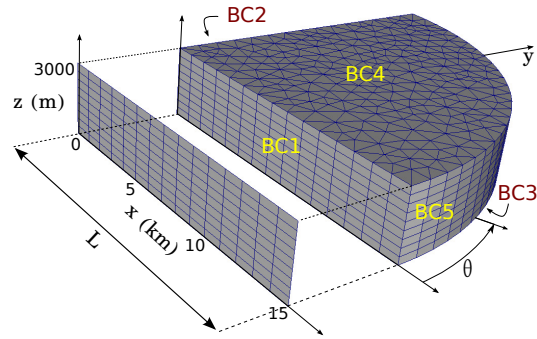


**Table 1.** Description and values of the model parameters.

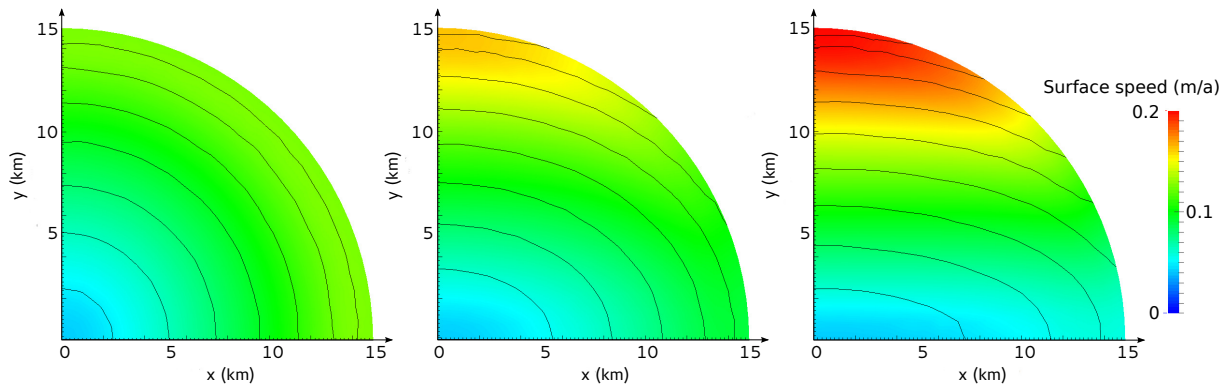
Parameter	Variable	Value	Unit
Accumulation rate	$a$	0.04	m/a
Flow line length	$L$	15 000	m
Downstream width	$W_L$	1	m
Flow law exponent	$n$	3.0	
Initial max. ice thickness	$\max(s)$	3239	m
Ice density	$\rho$	917	kg/m <sup>3</sup>
Mean surface slope		0.6 / 1000	



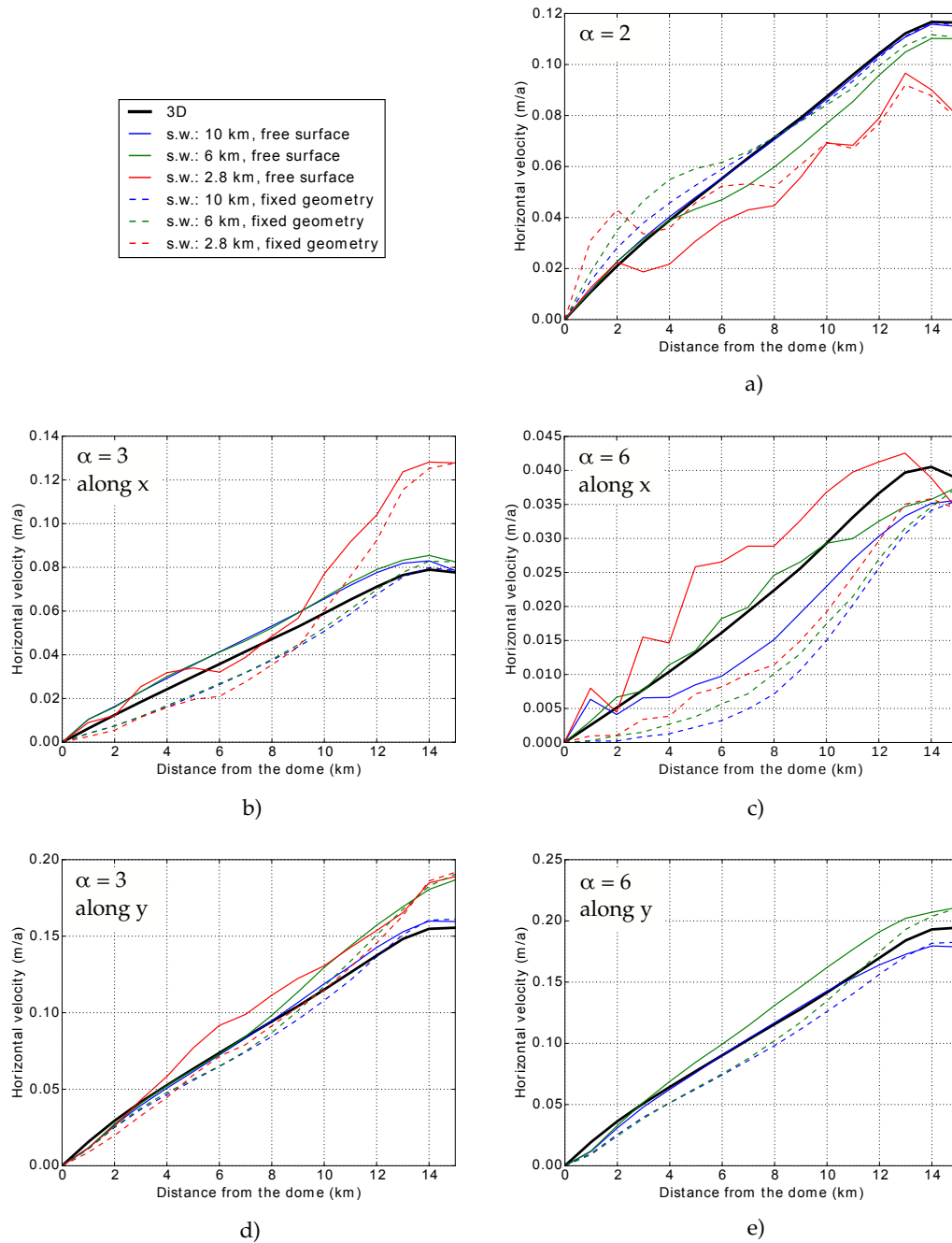
**Figure 1.** The  $x$ -axis is taken along the ridge stream line, and the  $y$ -axis tangent to the surface elevation contour. The scanning window is used to determine the value of  $R$  in its center. Lateral stream lines are represented to show the non linear widening of the flow tube in this particular case, and the corresponding accumulation surface is colored in gray.



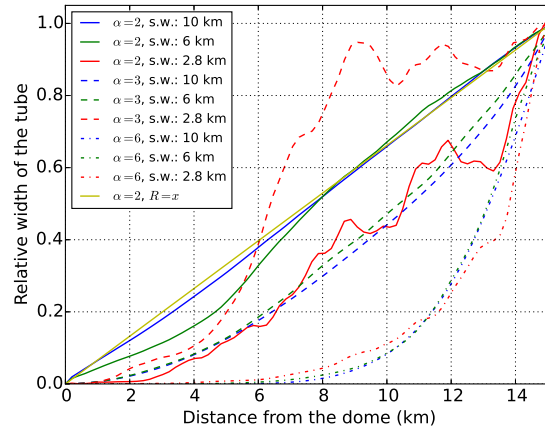
**Figure 2.** View of the two meshes used in this study. For each run, the 2D mesh is extracted from the geometry of the steady state solution of the 3D simulation. BC1 to BC5 refer to the 5 boundary conditions of the 3D case.



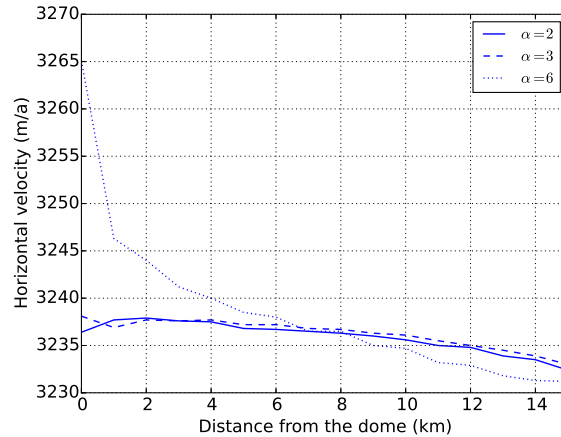
**Figure 3.** The three domes stabilized at steady state: surface contour lines (spacing: 1 m), and surface speed (m/a) for  $\alpha = 2$  (left),  $\alpha = 3$  (middle) and  $\alpha = 6$  (right).



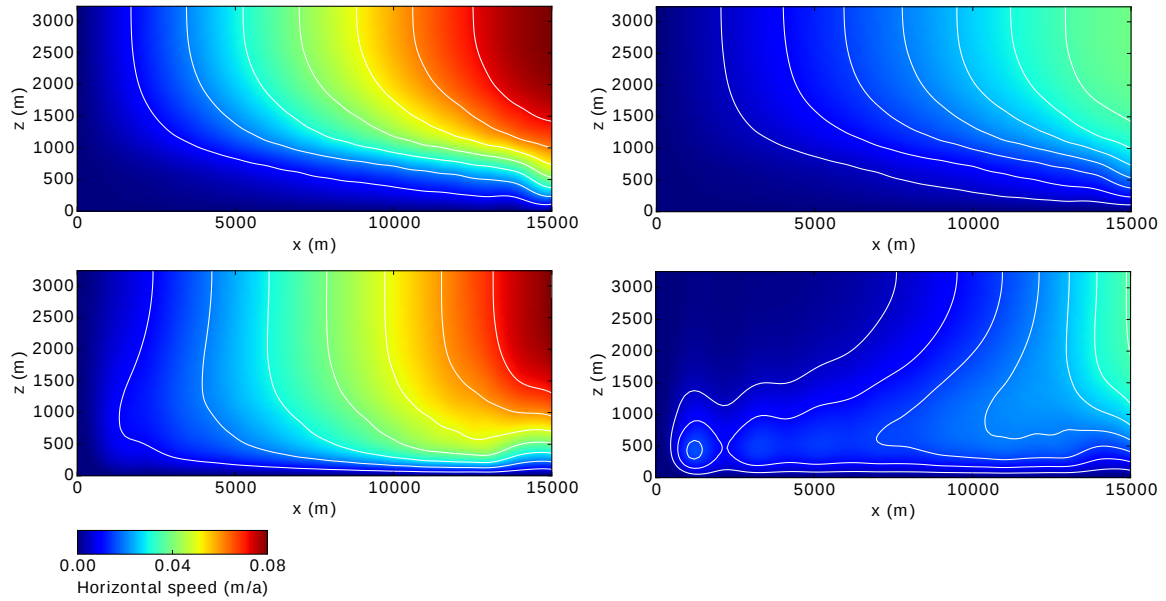
**Figure 4.** Horizontal velocity at the ice surface (m/a) along the flow line, for isothermal ice. Dashed lines are for fixed geometry runs, and solid lines for evolved free surface to a steady state. Thick black : 3D; yellow: 2.5D axisymmetric (hidden by the black curve); red: 2.5D scanning window of 2.8 km; green: 2.5D scanning window of 6 km; blue: 2.5D scanning window of 10 km. a) Circular dome,  $\alpha = 2$ . b)  $\alpha = 3$ , along the ridge. c)  $\alpha = 6$ , along the ridge. d)  $\alpha = 3$ , perpendicular to the ridge. e)  $\alpha = 6$ , perpendicular to the ridge. For this last case, no result concerning the 2.8 km scanning window is shown since they are completely out of reasonable bounds.



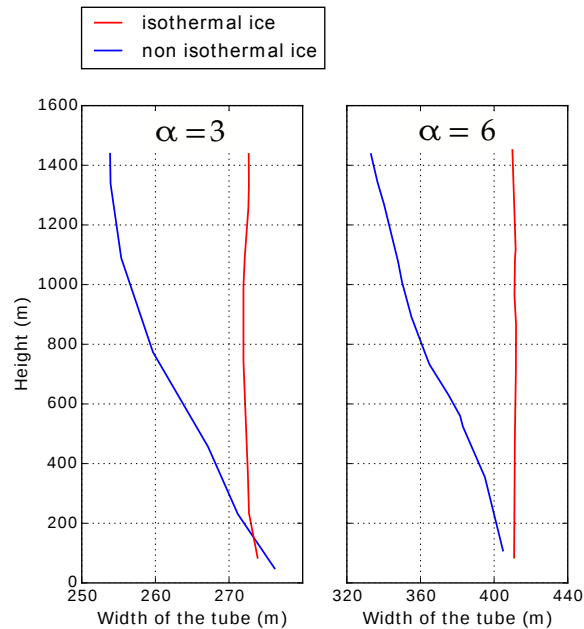
**Figure 5.** Relative width of the flow tube  $W$  for the different dome geometries: circular ( $\alpha = 2$ , solid lines), slightly elongated ( $\alpha = 3$ , dashed lines) and very elongated ( $\alpha = 6$ , dotted lines); for different scanning windows: 2.8 km (red), 6 km (green) and 10 km (blue). The yellow curve is the line for which  $R = x$  is imposed (axisymmetric case).



**Figure 6.** Height of the ice surface (m) for the 2.5D free surface model, along the ridge. Solid line: circular geometry,  $\alpha = 2$ . Dashed line:  $\alpha = 3$ . Dotted line:  $\alpha = 6$ .



**Figure 7.** Horizontal velocity field (m/a), for the non isothermal case and  $\alpha = 3$ . Top: 3D model. Bottom: 2.5D model, with a scanning window of 10 km. Left:  $\alpha = 3$ , white contour lines spaced by 0.01 m/a. Right:  $\alpha = 6$ , white contour lines spaced by 0.005 m/a.



**Figure 8.** Width of the flow tube (m), for the isothermal (red) and non isothermal case (blue), for  $\alpha = 3$  ( $x = 1500$  m, left) and  $\alpha = 6$  ( $x = 1200$  m, right). The aspect of the curves is slightly affected by numerical noise due to discretization.